

Toward Zero Emissions Electric Aircraft through Superconducting DC Distribution Network

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- **Hydrogen-powered Aircraft**
- **Cryogenic Power Electronics**
- **DC Network Protection Challenges**
- **Integration of SFCL and SSCB**
- **Summary**



**University of Bath
Institute for Advanced Automotive Propulsion (IAAPS)/
AAPS Centre for Doctoral Training**

- Delivery of fundamental research and innovation programs
- Funded industrial collaborative research projects
- Training the next generation of leaders and engineers to pioneer and shape the transition to clean, sustainable and affordable mobility



**IAAPS Limited
Wholly owned subsidiary of the University of Bath**

- Commercial focused delivery of research, innovation and testing programmes for industry
- Supporting the University of Bath in delivery research and innovation programmes
- Regional enterprise activity

Introducing Airbus ZEROe

Turboprop



<100
Passengers



Hydrogen
Hybrid Turboprop
Engines (x 2)



1,000+nm
Range



Liquid Hydrogen
Storage & Distribution
System

Blended-Wing Body



<200
Passengers



Hydrogen
Hybrid Turbofan
Engines (x 2)



2,000+nm
Range



Liquid Hydrogen
Storage & Distribution
System

Turbofan



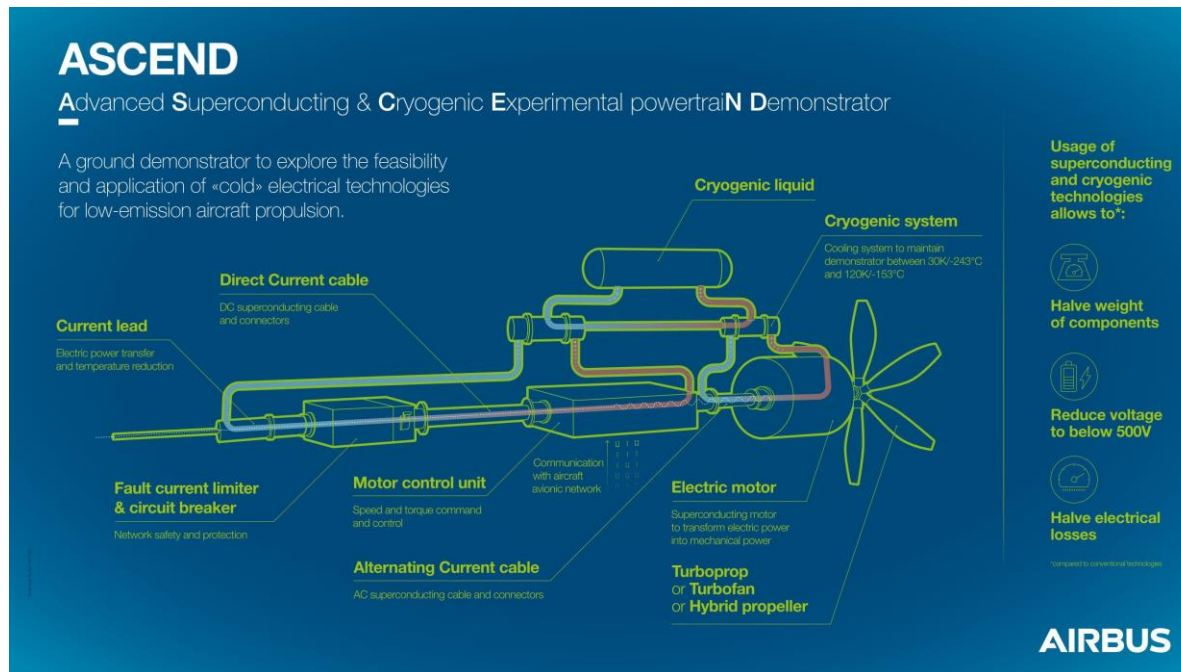
AIRBUS

Airbus' ambition is to bring to market the world's first hydrogen-powered commercial aircraft by 2035.

Source: Airbus

Hydrogen-powered Aircraft

- Zero emission aircraft powered by hydrogen address the environment impact and also opens new opportunities for superconductivity technology.
- Superconducting and cryogenic propulsion system has great potential to deliver high power density and high efficiency.



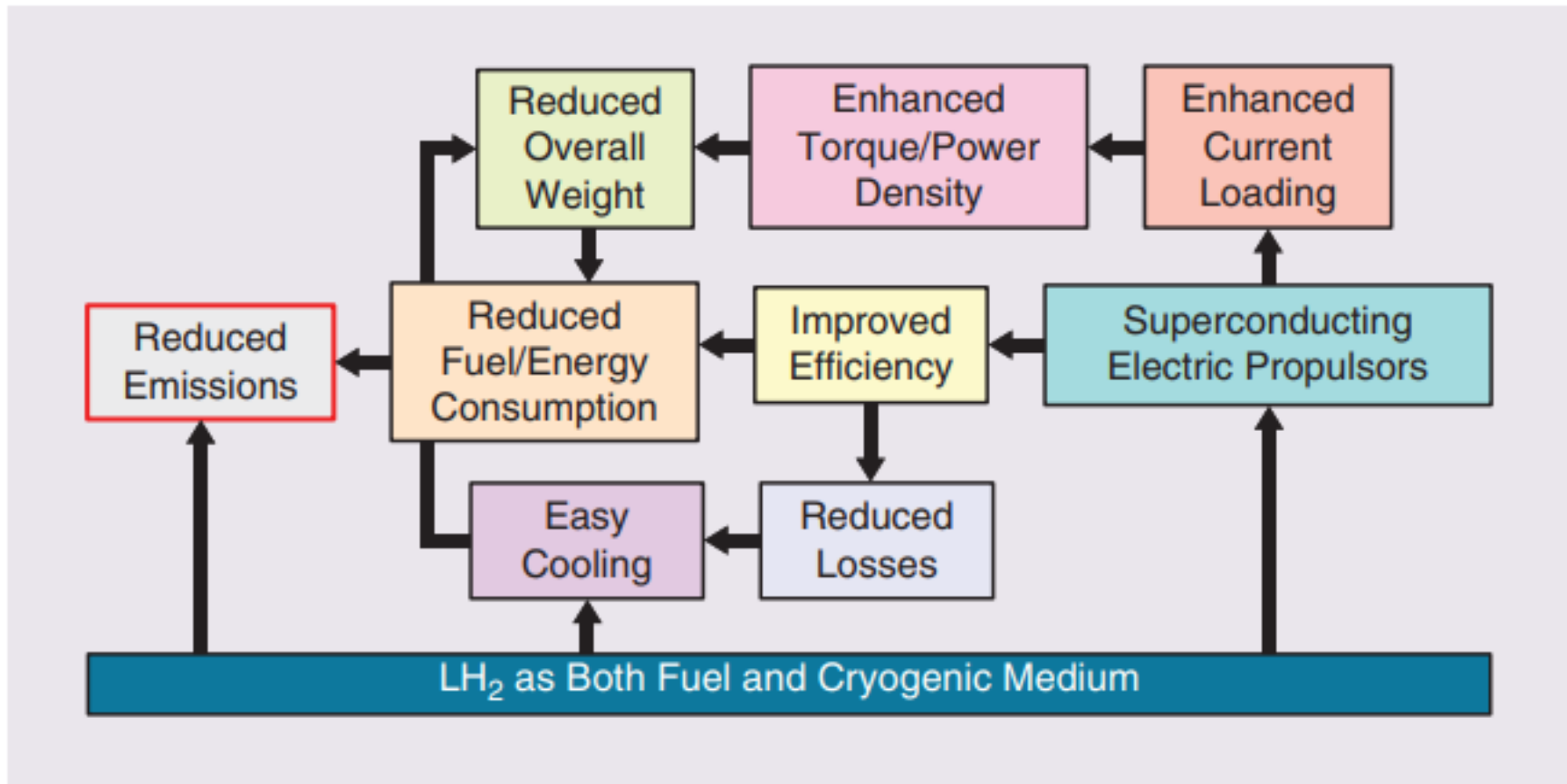
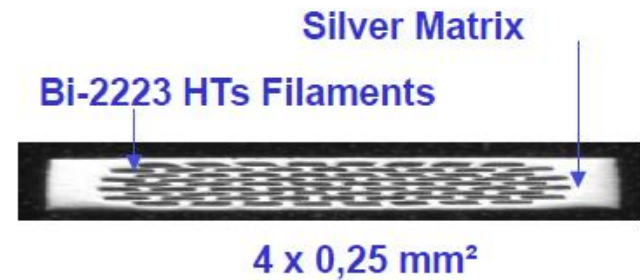


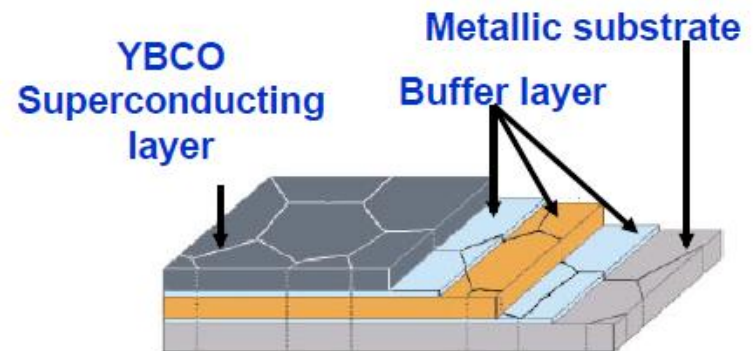
Figure 12. The overall philosophy when reducing emissions using LH₂ as a fuel and a cryocooler.

High Temperature Superconductor Materials

- 1st generation: Multifilamentary Bi 2223 tapes

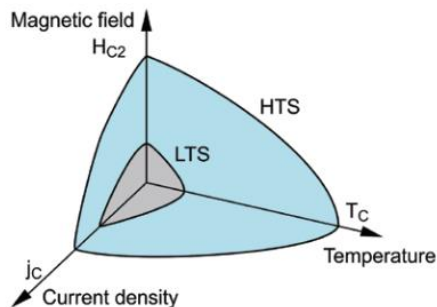
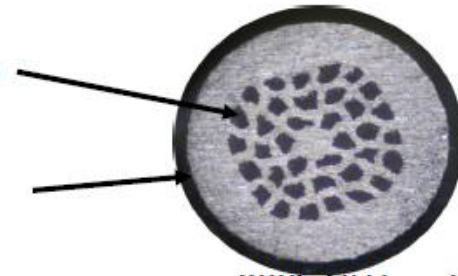


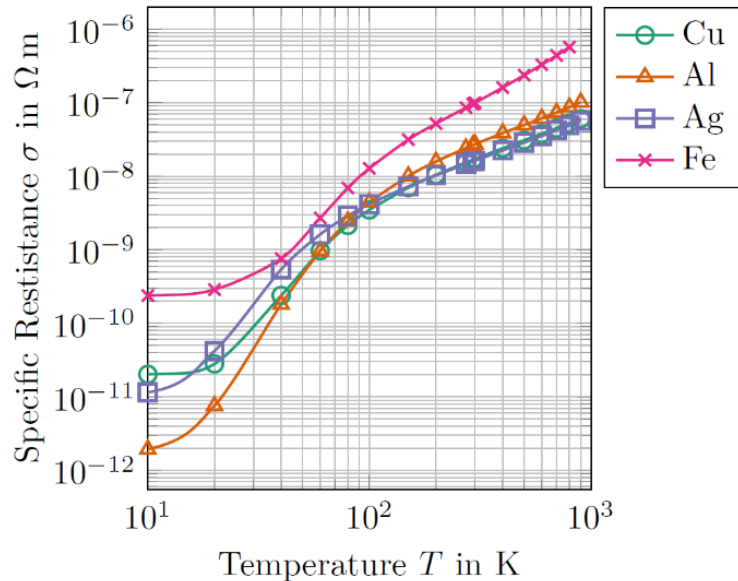
- 2nd generation: YBCO Coated conductor



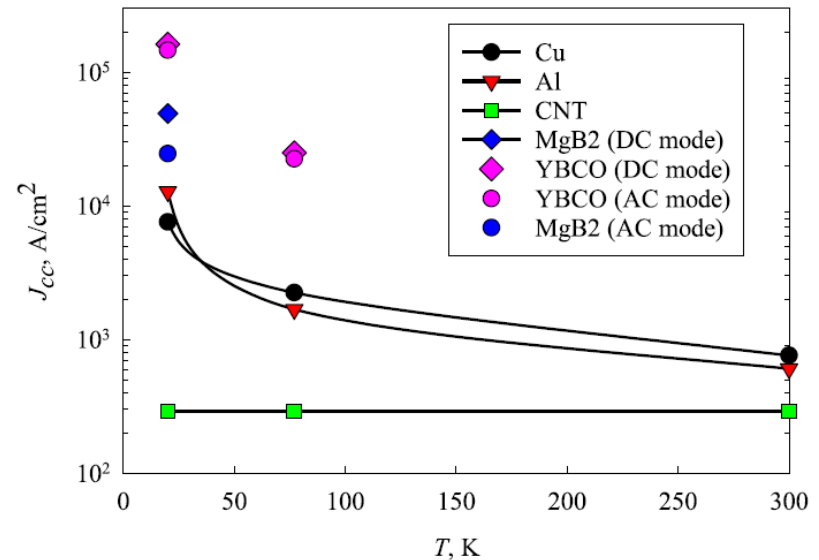
- Multi-filamentary wire ($T_{op} = 20-25K$) MgB_2 Filaments

Copper and nickel matric





Specific Resistance vs Temperature



Current Density vs Temperature

Source: David Filusch, Martin Breiteneder and Hans-Georg Herzog, "Design of a Hyperconducting Synchronous Machine for High-Torque Applications".

M.D. Sumption, J. Murphy, M. Susner, T. Haugan, "Performance metrics of electrical conductors for aerospace cryogenic motors, generators, and transmission cables, Cryogenics, Volume 111, 2020, 103171.

Hydrogen-powered Aircraft



Collaborating to develop a ground-breaking hydrogen propulsion system powering
Zero Emissions Aircraft

H₂GEAR HOW IT WORKS



In partnership with:



Global Technology Centre Bristol

THE BENEFITS

£54 M+
initial investment
(£27.2M Gov, matched by industry)

FURTHER
£200M+
R&D investment
to follow

UNLOCKING
a market worth
£16.8Bn

3,120
jobs created in the
UK

ZERO
CO₂ emissions
air travel

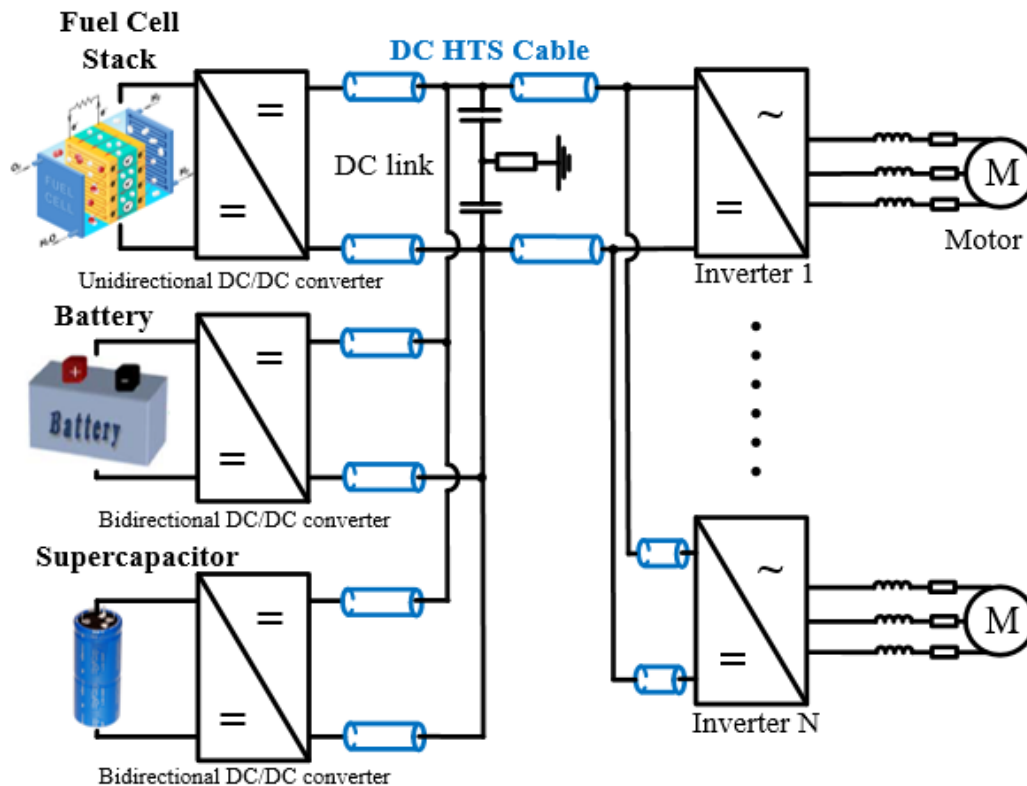
100%
UK system
content

2026 First potential
entry into
service 

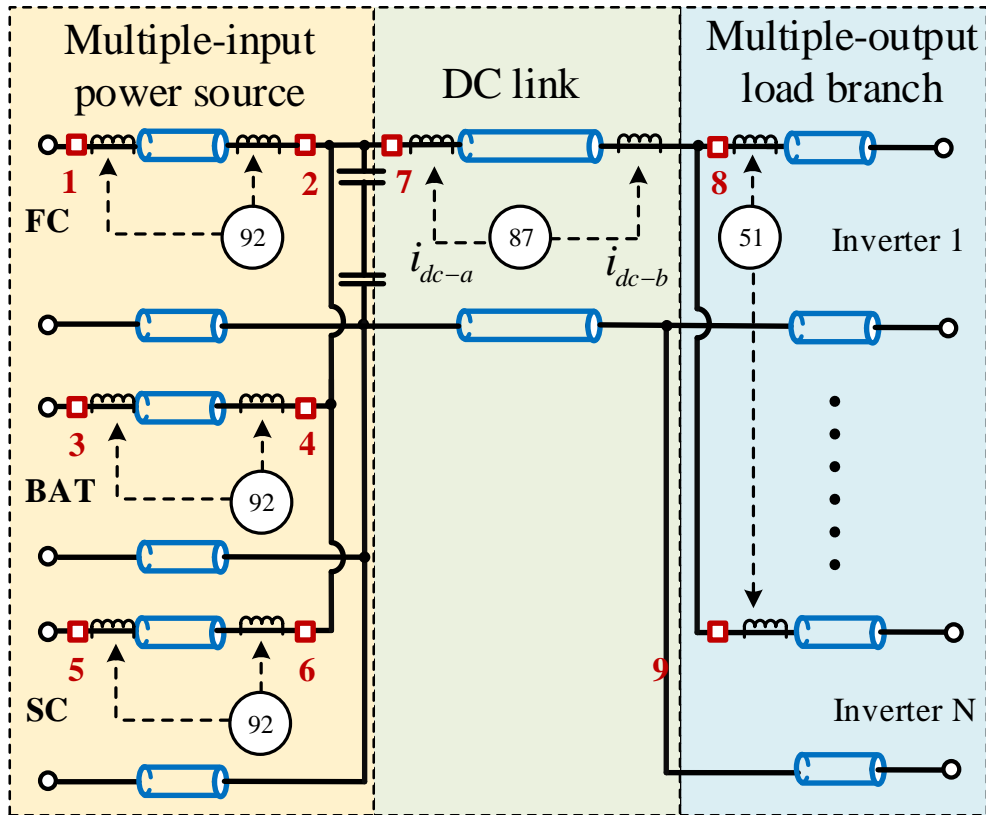
“ Working with our partners, and made possible by Government investment, GKN Aerospace will develop and industrialise the breakthrough technology to fly aircraft with zero CO₂ emissions by the mid-2020s. ”

Russ Dunn, CTO GKN Aerospace

Fuel Cell Propulsion System



- DC/DC/AC system
- DC/DC converters as power controller
- Fuel cell as primary power source
- Batteries and supercapacitors as energy storage system
- DC link voltage regulated by batteries and supercapacitors

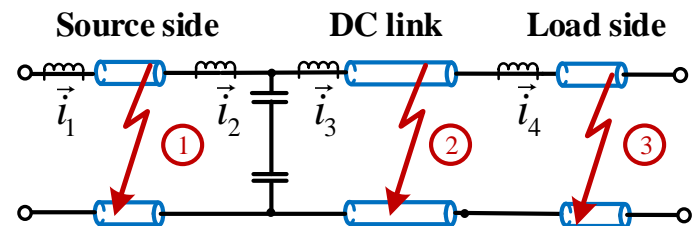


⌘: Current transformer ◻: Circuit breaker (CB)

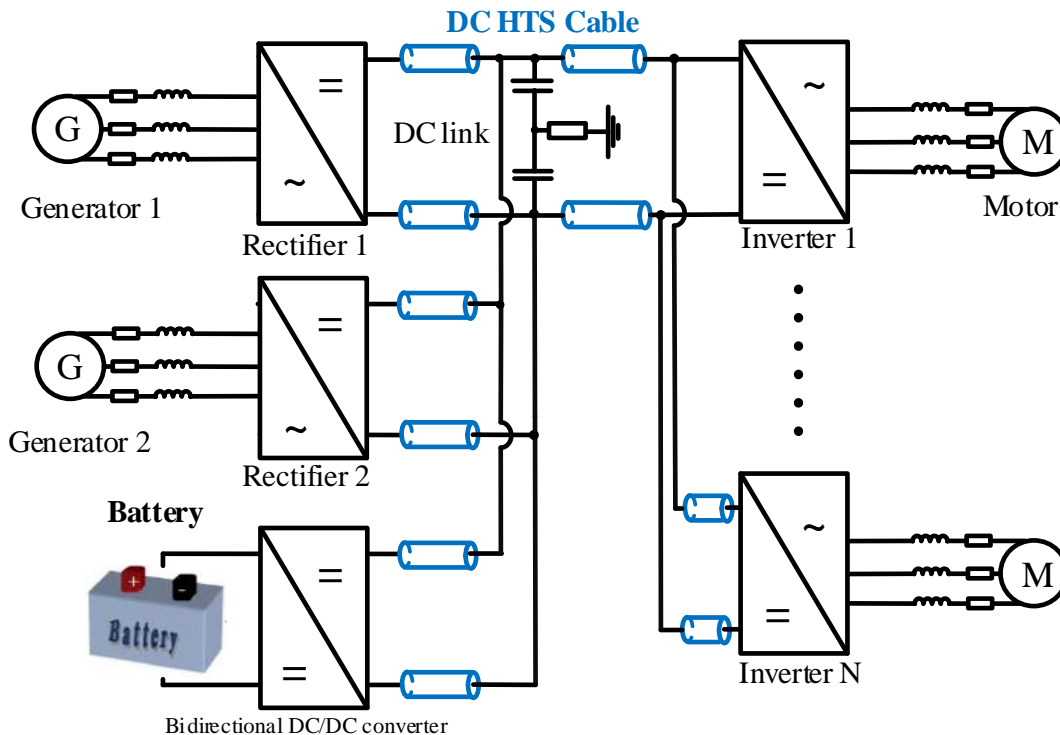
⊙(87): Differential relay ⊙(92): Directional relay ⊙(51): Overcurrent relay

Potential fault locations and types

1. Power source side fault
 - Ground fault + Short-circuit fault
2. DC link side fault
 - Ground fault + Short-circuit fault



Turbo Electric Propulsion System



- AC/DC/AC system
- Distributed generations and distributed loads
- Converter based power network
- DC link voltage regulated by rectifiers

Power devices at cryogenic temperatures compared with room temperature

Type	Si MOSFET	Si IGBT	GaN HEMT	SiC MOSFET
On-state resistance	Lower	Lower	Lower	Higher
Switching loss	Lower	Lower	Higher	Constant
Breakdown voltage	Lower	Lower	Constant	Constant
Threshold voltage	Higher	Higher	Constant	Higher

Cryogenic Power Electronics

Si MOSFET at cryogenic temperatures

- Lower on-state resistance
- Lower switching loss
- Reduced breakdown voltage

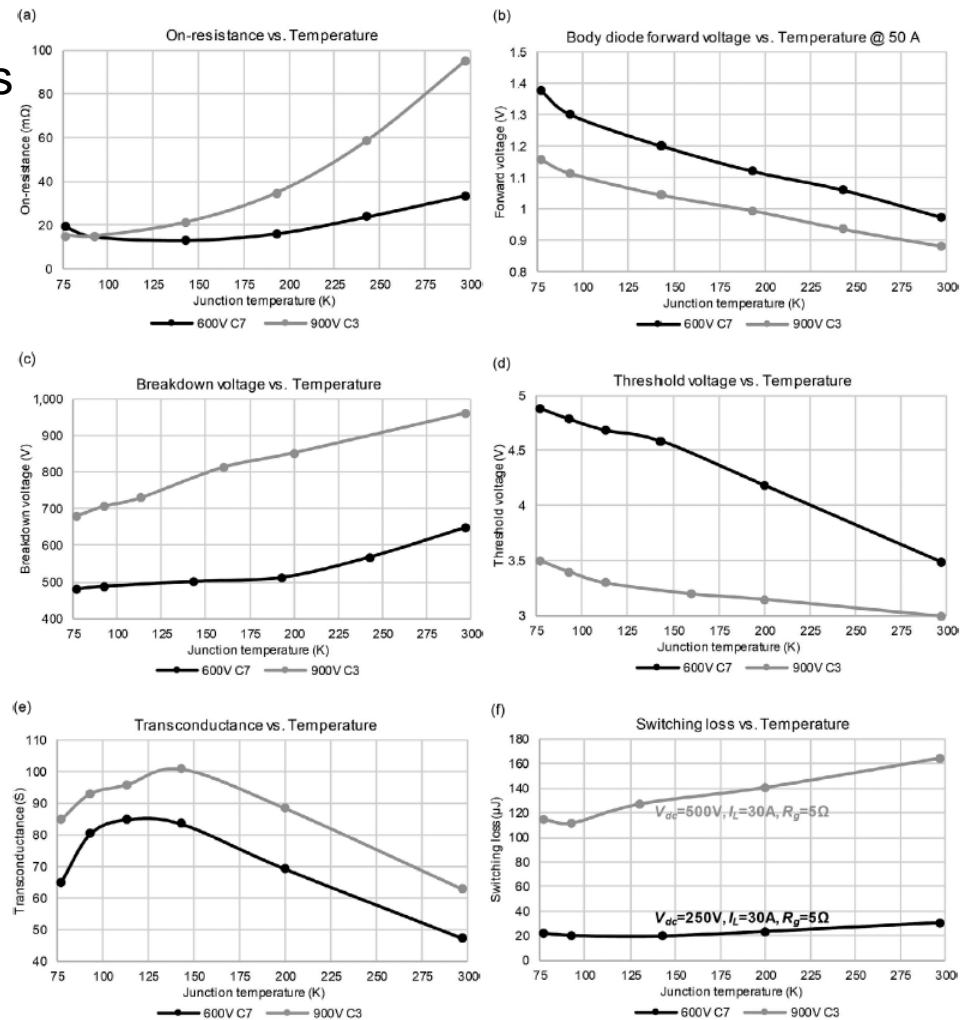


Figure 6.6 Performance of two Si MOSFET candidates at cryogenic temperature. © [2020] IEEE. Reprinted, with permission, from [26]

Passive components

COMPARISON OF DIFFERENT CAPACITORS AT CRYOGENIC TEMPERATURES WITH ROOM TEMPERATURE

Material	Ceramic				Film				Mica	Electrolytic	Tantalum
	X7R	Y5V	Z5U	NP0	Polypropylene	PPS	Polyester	Polycarbonate			
Capacitance	↓↓	↓↓	↓↓	--	--	--	↓	↓	--	↓↓	↓
Dissipation factor	↑↑	↑↑	↑↑	--	↓	--	↓	↓	--	↓↓	↑↑

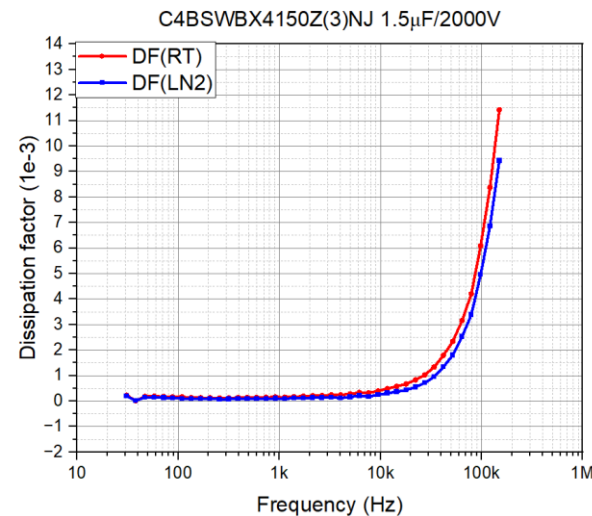
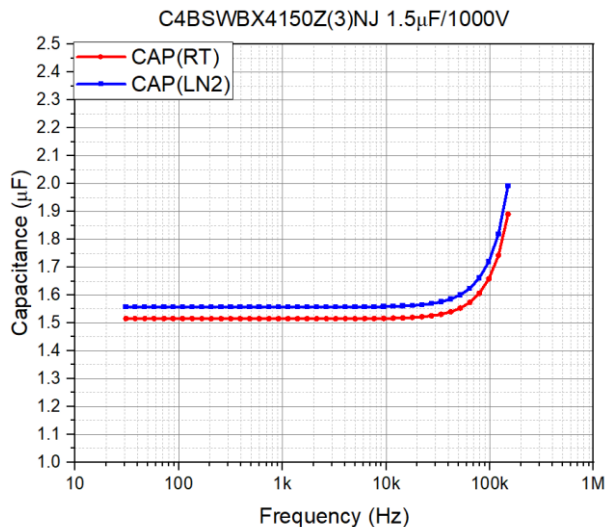
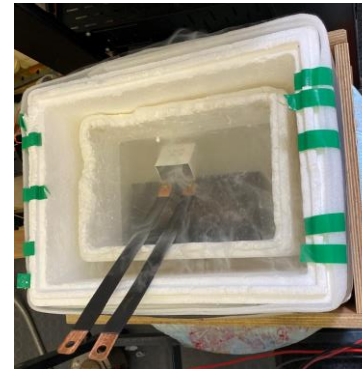
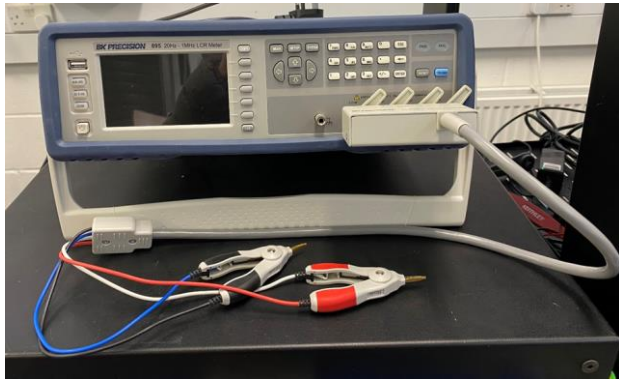
- NP0, metallized polypropylene, mica, and polyethylene capacitors maintain capacitances and/or have improved dissipation factors at lower temperatures.

COMPARISON OF DIFFERENT CORE MATERIALS AT CRYOGENIC TEMPERATURES WITH ROOM TEMPERATURE

Material	Powder			Ferrite	Nano-crystalline	Amorphous
	MP	High Flux	Kool M μ			
Permeability	--	--	↓	↓↓	↑	--
Loss	↑	--	--	↑↑	↑	↑
Saturation flux density	N/A	N/A	N/A	Not clear	↑	↑

↑↑: increase significantly ↑: increase slightly --: keep constant ↓: decrease slightly ↓↓: decrease significantly

Capacitor characterisation



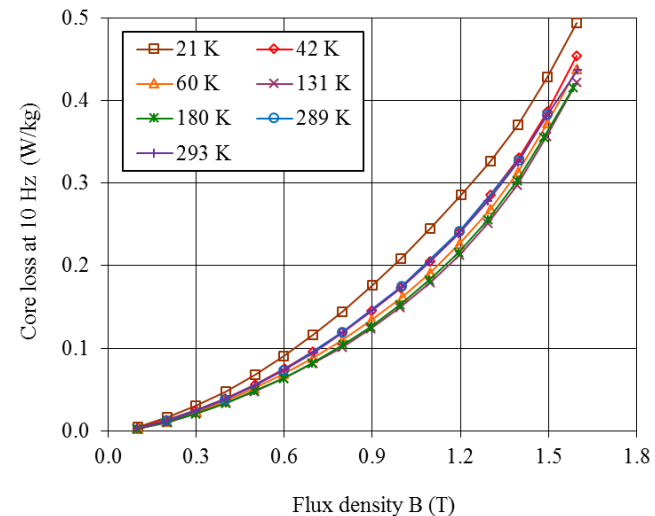
Magnetic core characterisation



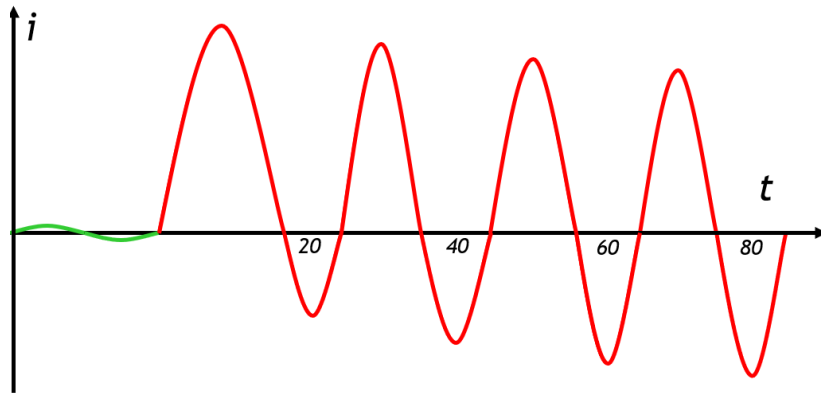
Fig. 1. Picture of soft magnetic ring core samples: (a) fully-processed, low loss non-oriented steel; (b) fully-processed, non-oriented steel with enhanced permeability; (c) grain-oriented steel; and (d) nano-crystalline alloy Finemet.

MAGNETIC PROPERTIES OF FOUR RING CORE SAMPLES

Sample	Temperature of best magnetic properties (K)	Core losses at best temperature/room temperature	Core losses at 21 K/room temperature
Low loss NO steel	131	0.93	1.12
NO steel with enhanced permeability	180	1	1.4
GO steel	289	1	1.2
Finemet	289	1	-



- Electric aircraft has high reliability and safety requirements.
- Potentially very high fault current as the on-board power network is closely coupled.



AC

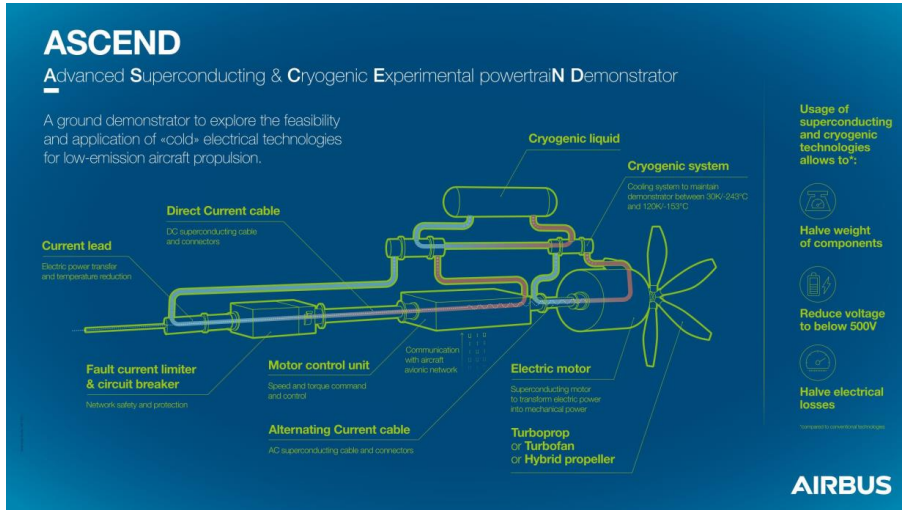
- Current crosses zero twice in each cycle.



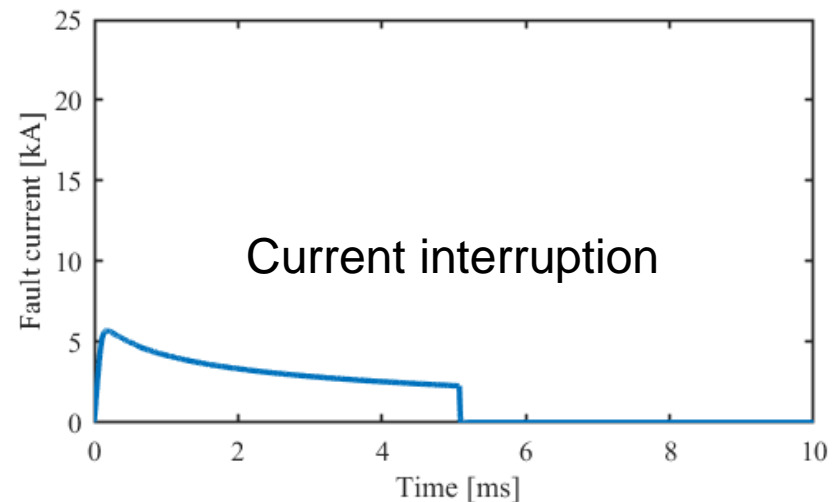
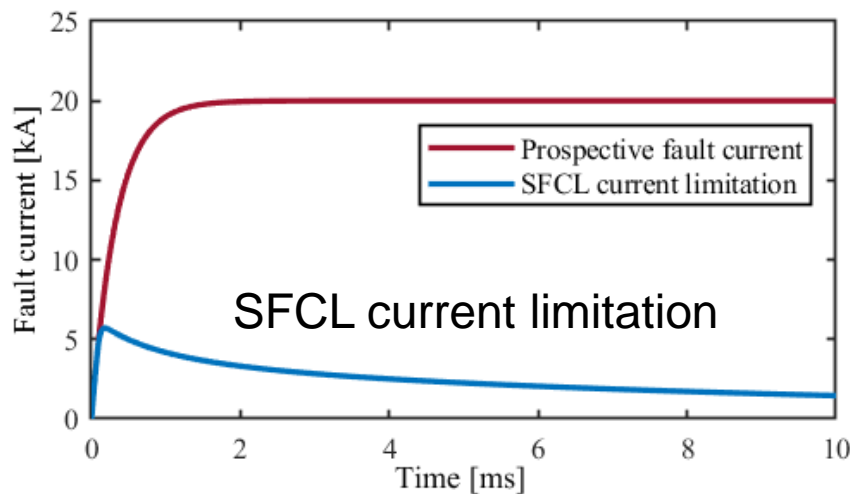
DC

- No natural current zero.
- High fault current level.
- High rate of rise of fault current.

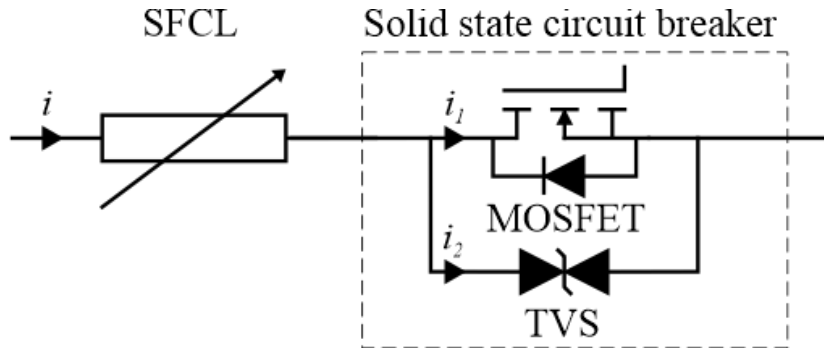
ASCEND Specification



Power rating	500 kW
Nominal voltage	300 V
Nominal current	1700 A
Prospective fault current	10-20 kA
Maximum limited fault current	6800 A
Withstand voltage	2000 V
Fault interruption time	5-10 ms

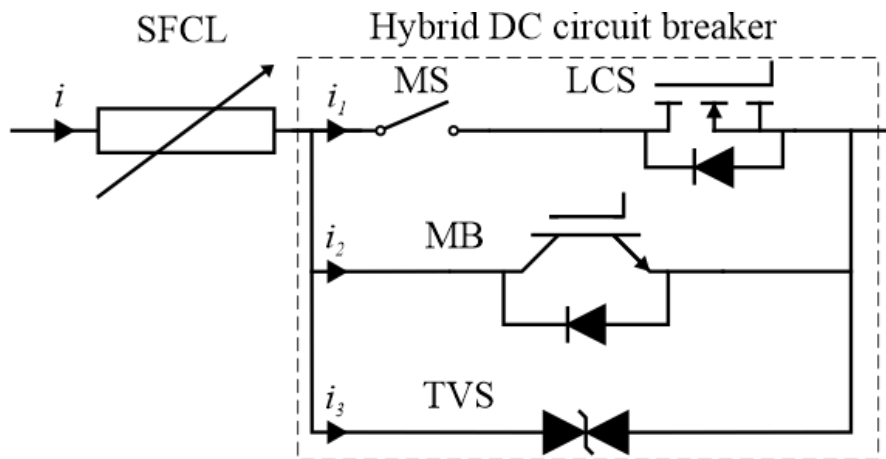


Technical solution 1: SFCL and solid state circuit breaker

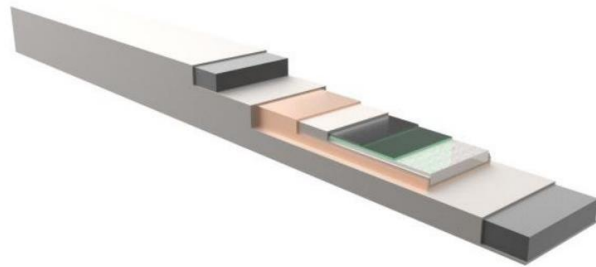


- Faster interruption

Technical solution 2: SFCL and hybrid DC circuit breaker

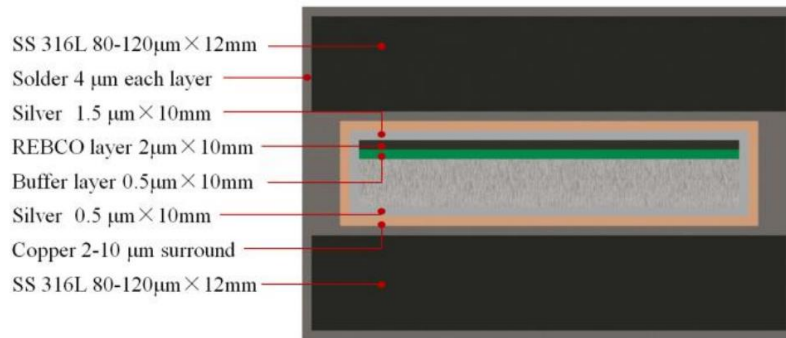
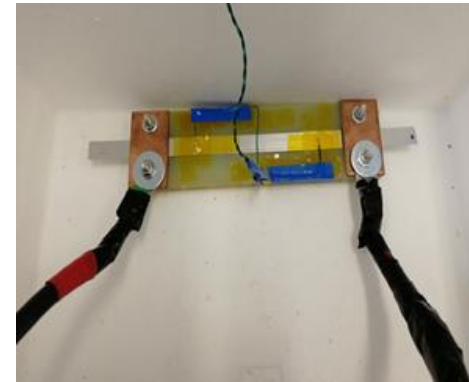


- Lower loss



(a)

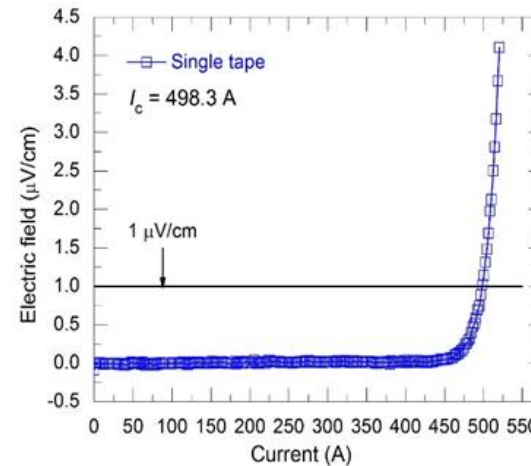
A



(b)

B

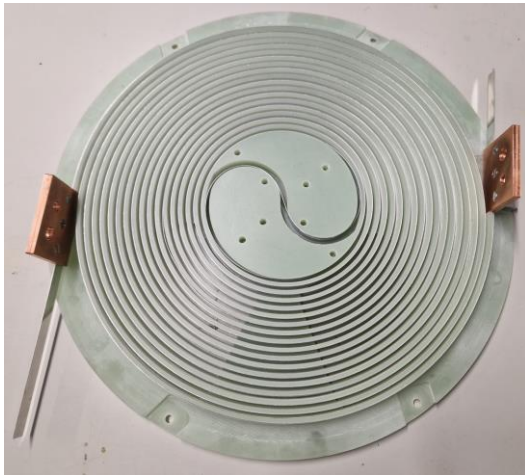
Short sample



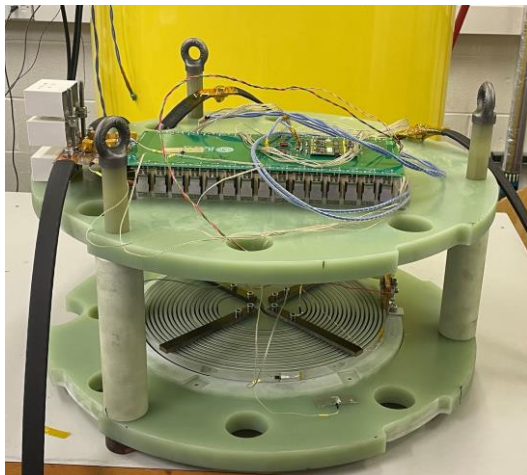
Shanghai superconductor 2G HTS tape
with stainless steel lamination

Critical current measurement

Superconducting Fault Current Limiter



SFCL pancake coil



SFCL on the sample holder

Items	Values
Tape	ST-12-L
Lamination material	Stainless steel
Tape width	12 mm
Tape thickness	0.25 mm
Resistance @ RT	100 mΩ/m
Critical current @ 77K	533 A
Critical current @ 65K	1023 A
Tape length	11.6 m
Number of turns	10×2
Diameter of former	400 mm

J. Hu, J. Xi, Z. Wang, X. Zeng, E. Nilsson, J. Rouquette, L. Ybanez and X. Pei, "Preliminary Design of DC Resistive Superconducting Fault Current Limiter for ASCEND," IEEE Transactions on Applied Superconductivity, vol. 34, no. 3, pp. 1-5, May 2024, Art no. 5600305, doi: 10.1109/TASC.2023.3338134.

Applied Superconductivity and Cryogenic Propulsion Laboratory

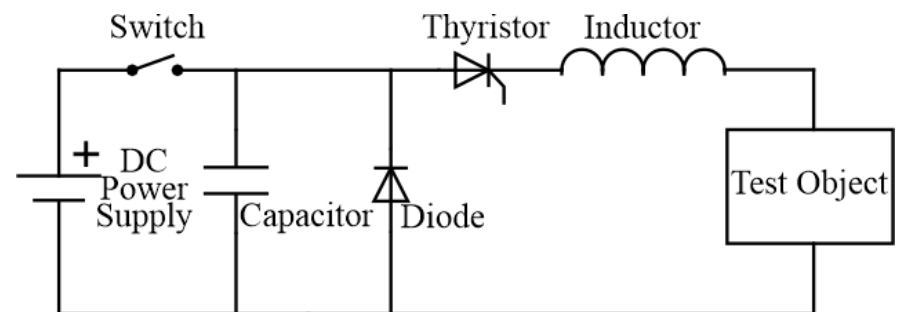
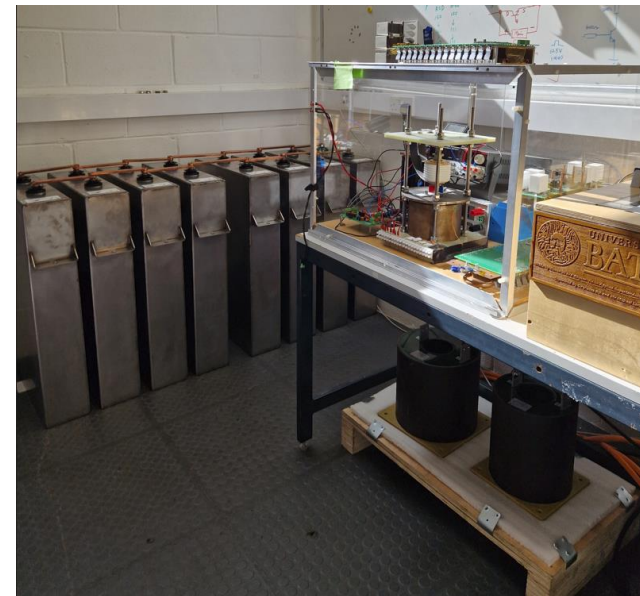
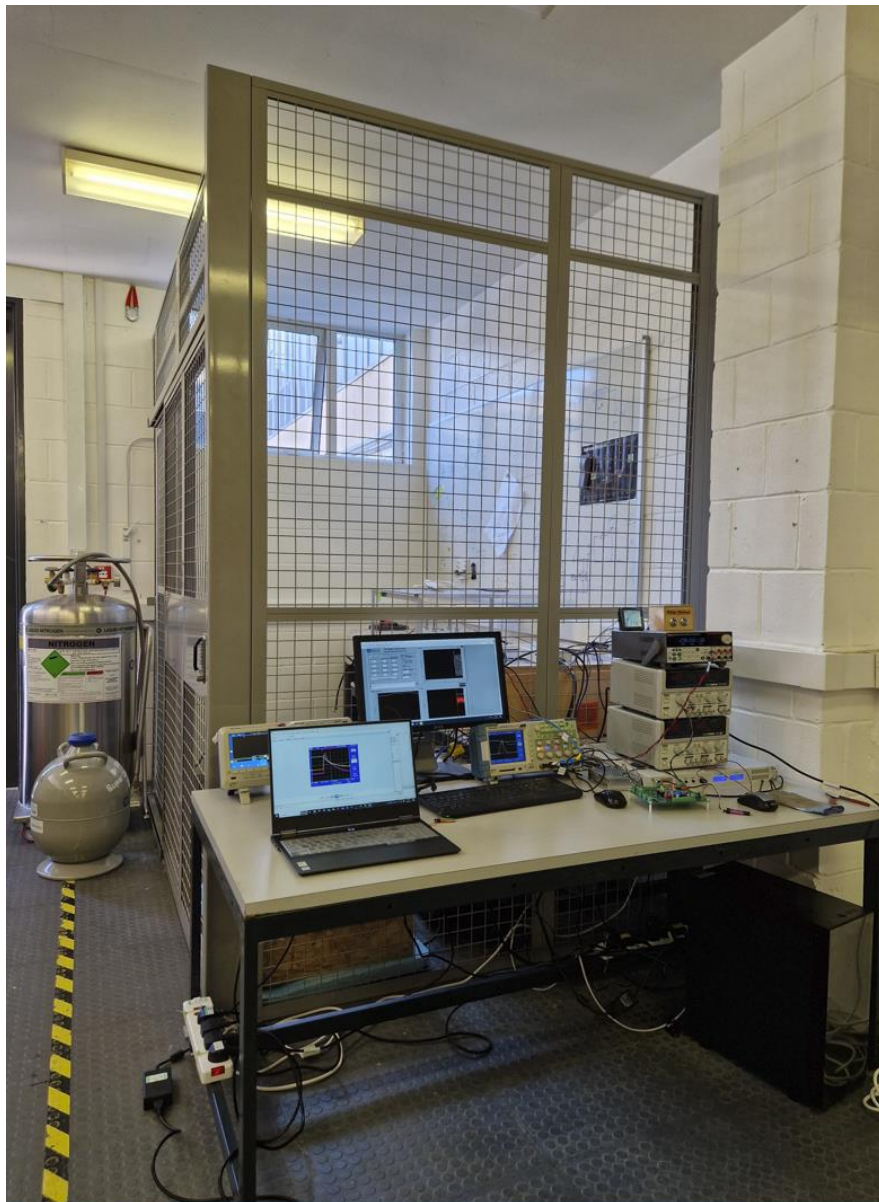


Cryogenic Cooling System



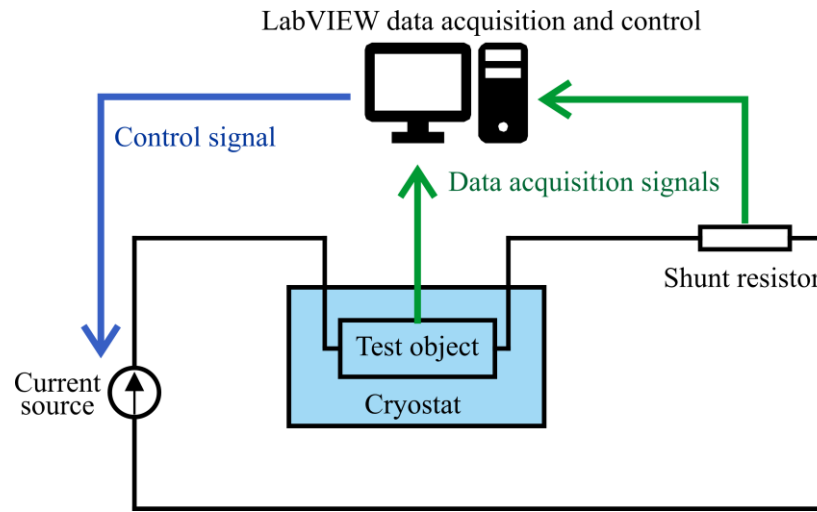
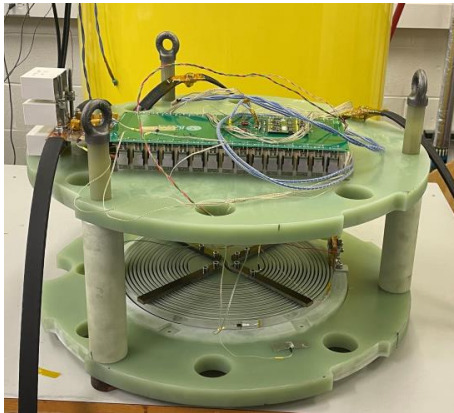
- Closed loop cryostat controllable from RT to 35 K
- Cryomech AL600 cold head
- Four current leads of 1000A, 2000V

DC Fault Current Test Rig



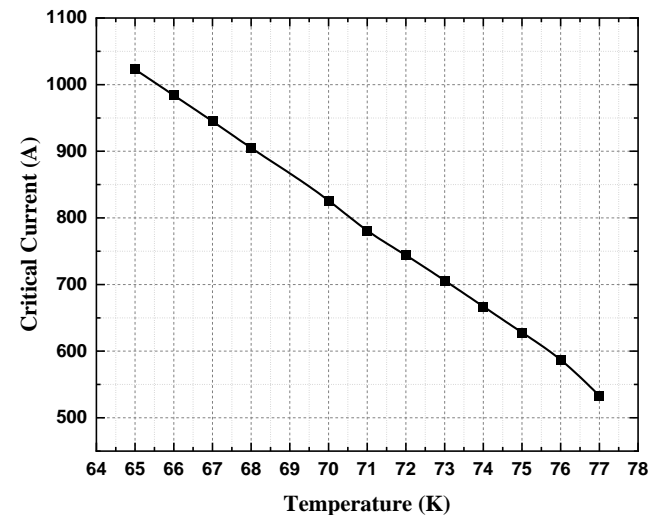
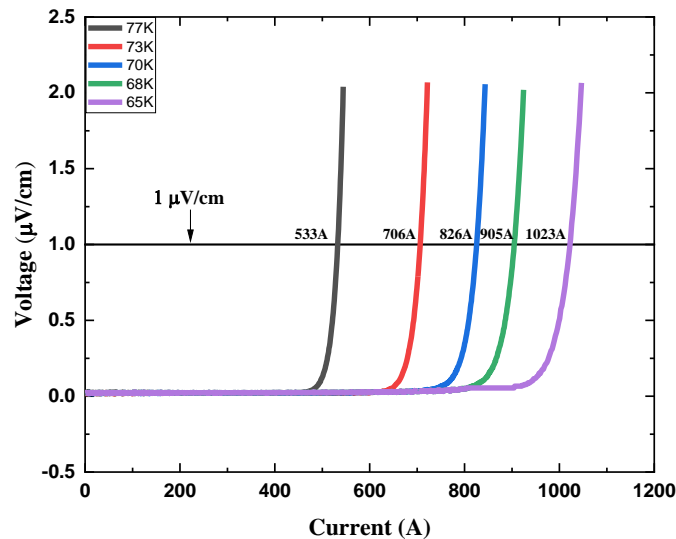
- Automatic control with data acquisition to simulate DC fault current up to 20 kA.

SFCL Experimental Test Results



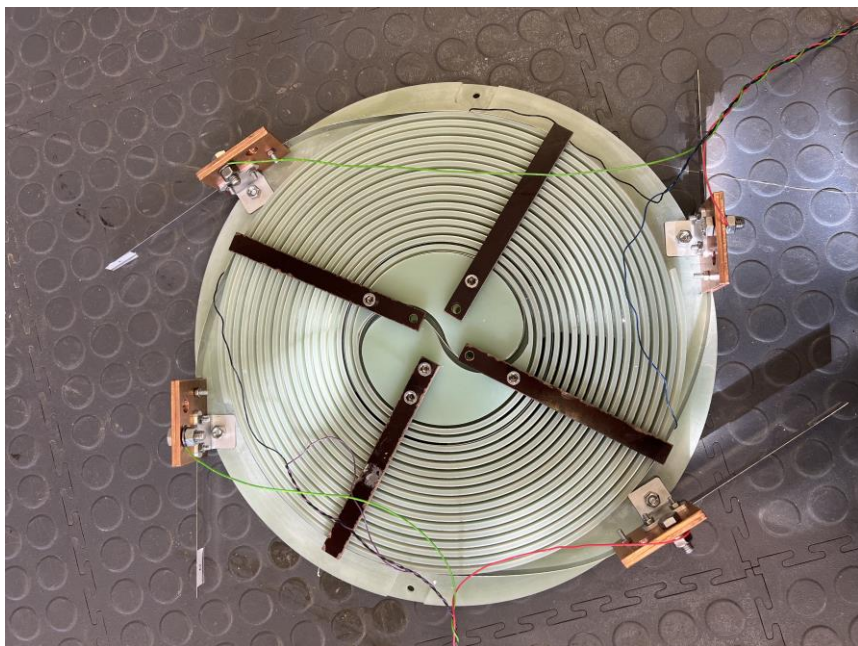
SFCL on the sample holder

SFCL coil critical current test circuit

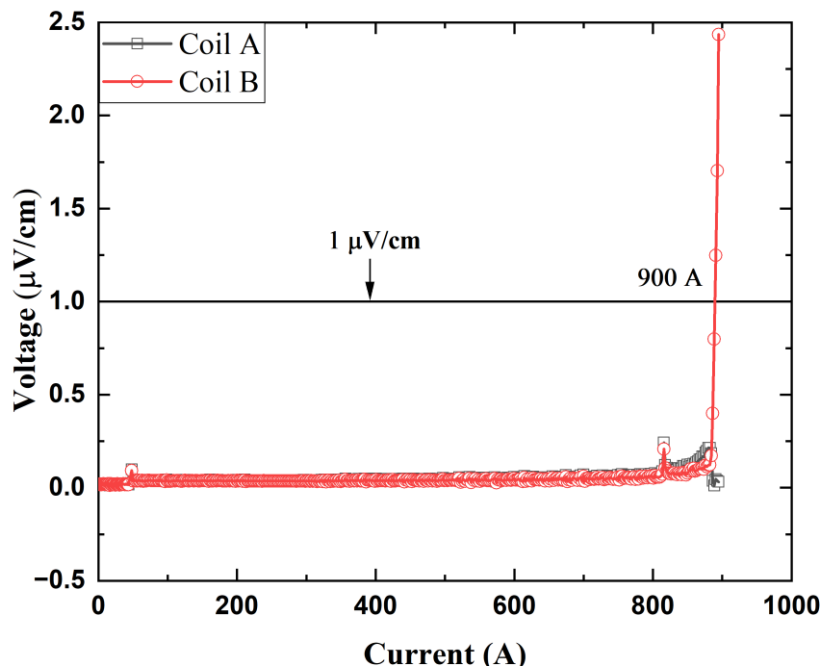


SFCL coil critical current

Double Tape SFCL

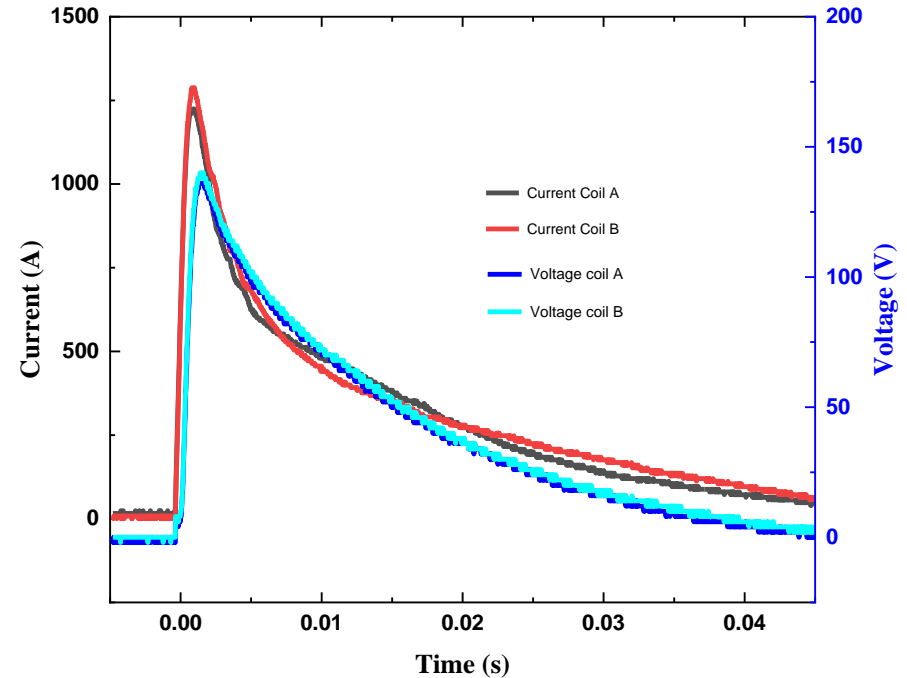
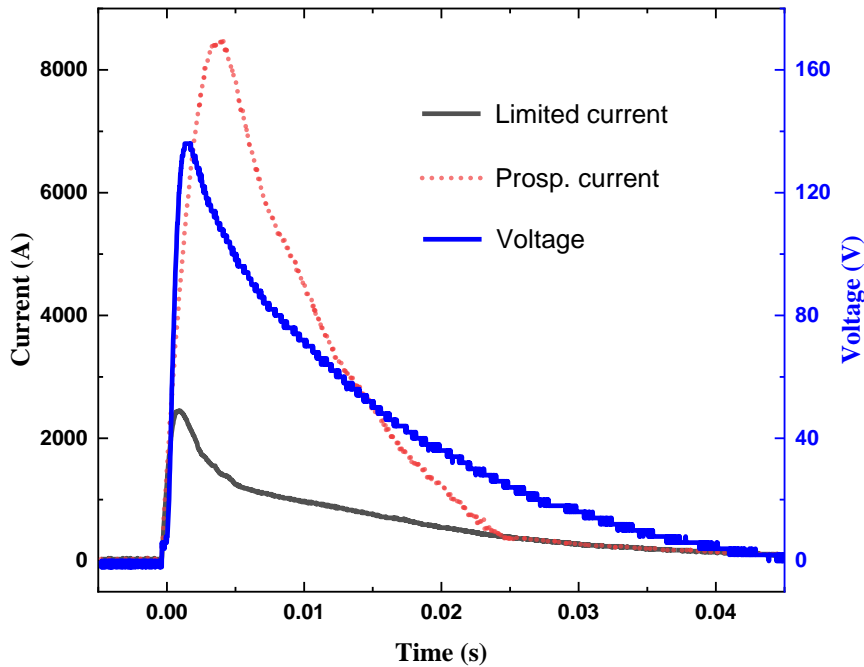


SFCL made by two tapes in parallel



Critical current test at 77 K

Double Tape SFCL Experimental Test Results

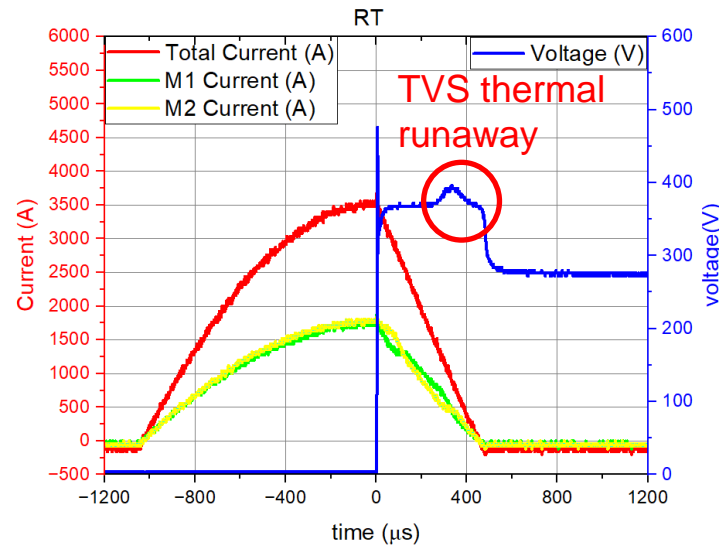
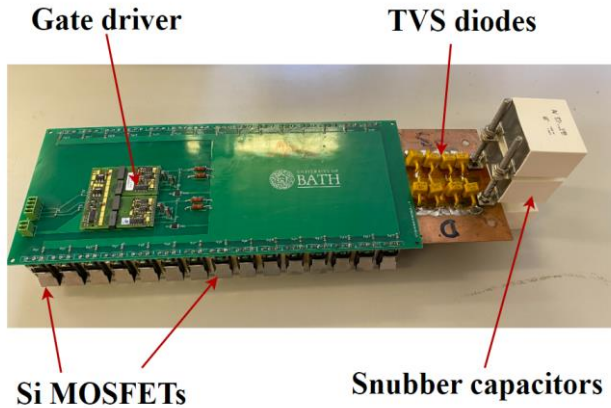


Double tape SFCL current limitation

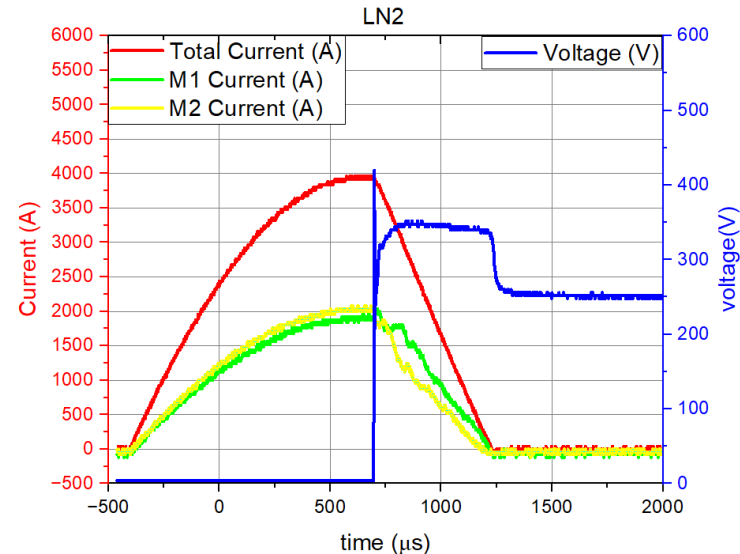
Current sharing in two HTS tapes

- Even current sharing and fault current limitation in two parallel HTS tapes.
- SFCL limits the fault current to 2-3 times critical current.

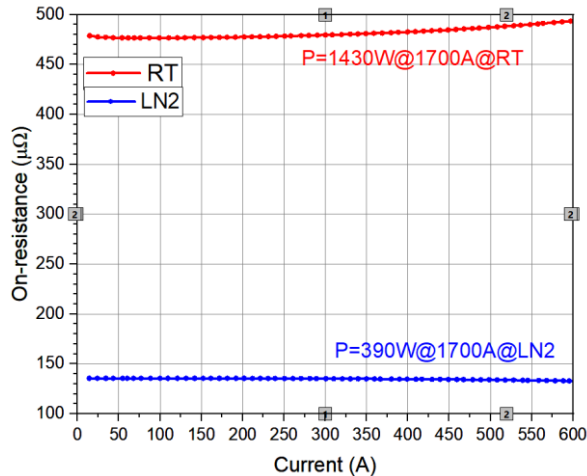
Solid State Circuit Breaker



Current interruption at RT

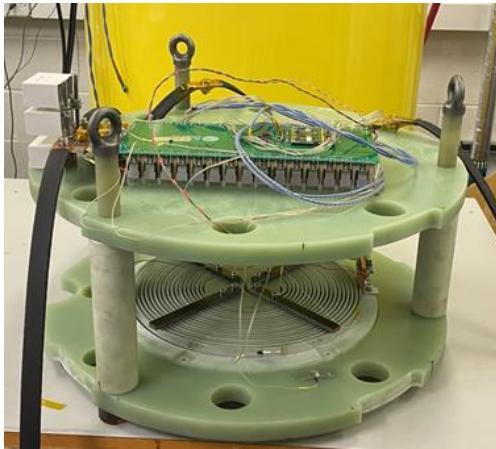


Current interruption in liquid nitrogen

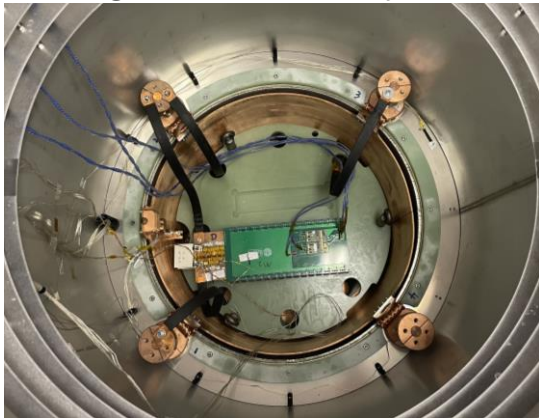


- SSCB on-state losses is reduced to 25% in liquid nitrogen than room temperature.
- TVS diodes are used absorb residual energy and liquid nitrogen provides better thermal condition than air.

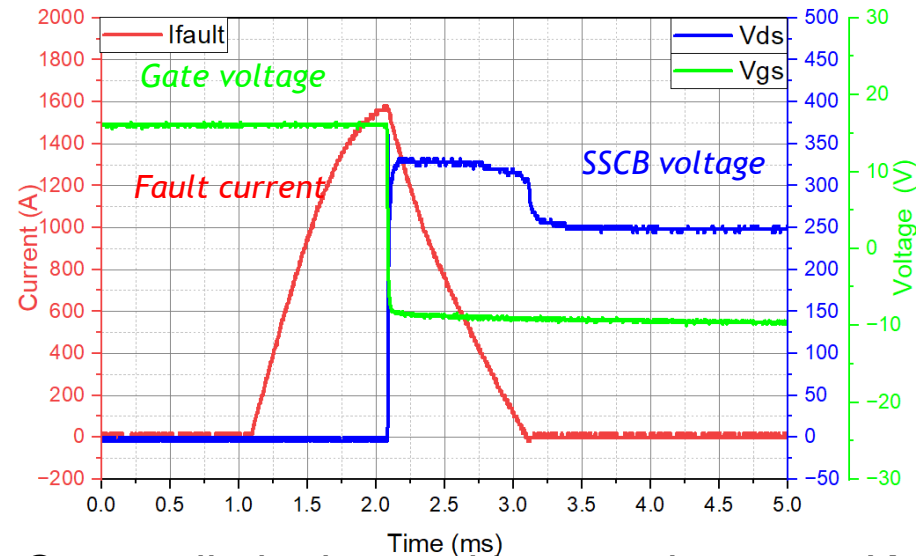
Integration of SFCL with SSCB



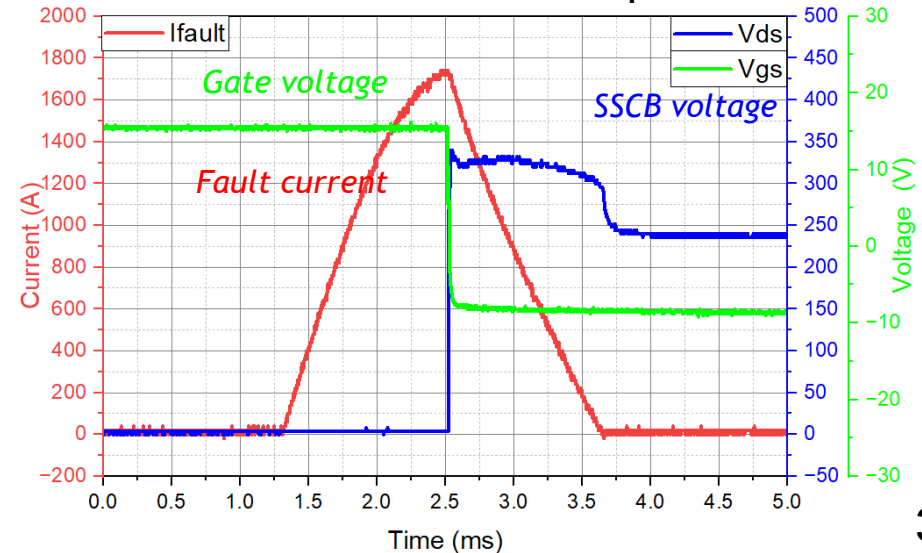
Integrated into cryostat



- SFCL and SSCB successfully operate in both LN2 and subcooled LN2.
- SFCL coil limits fault current from 2.5 kA to 1.6 kA at 77 K, and SSCB successfully interrupts the fault current at 1.6 kA.



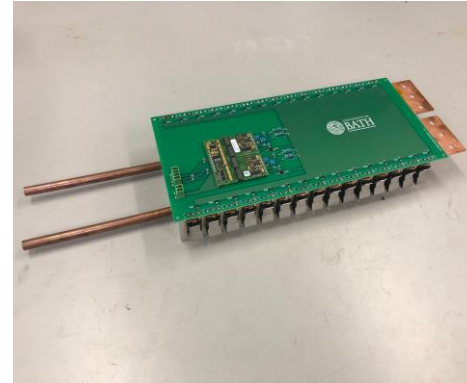
Current limitation and interruption at 77 K



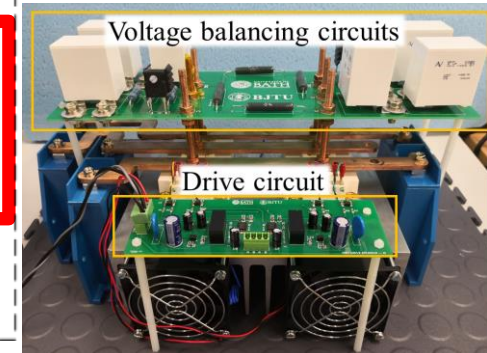
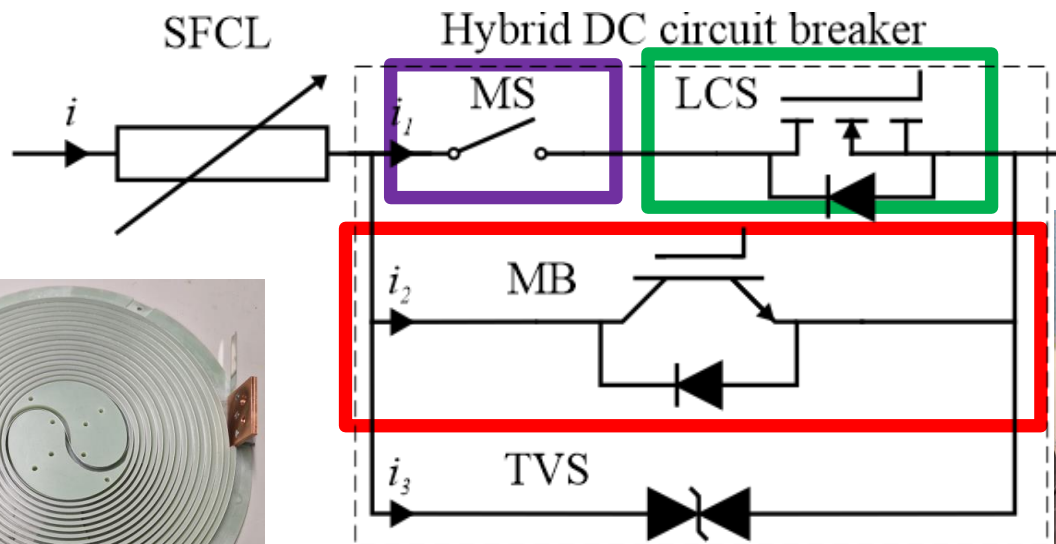
Current limitation and interruption at 73 K

Integration of SFCL with HCB

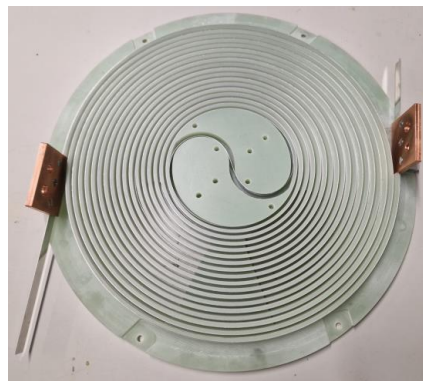
Fast operating
mechanical switch



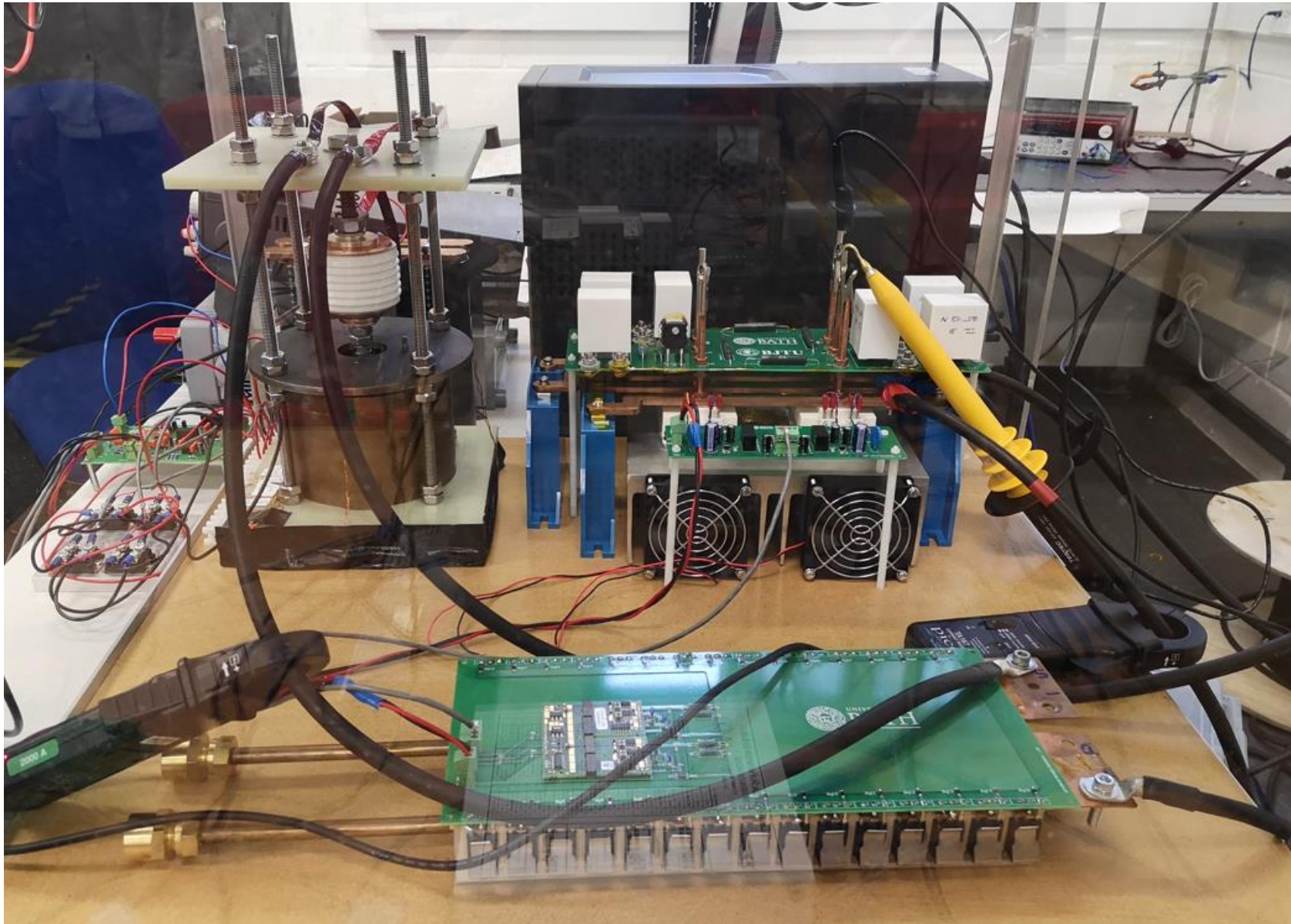
Low loss load
commutation
switch



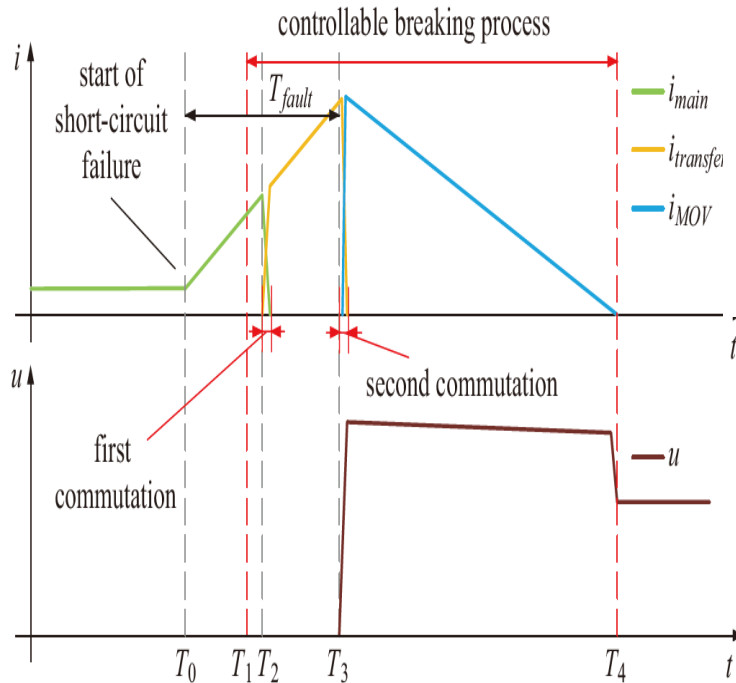
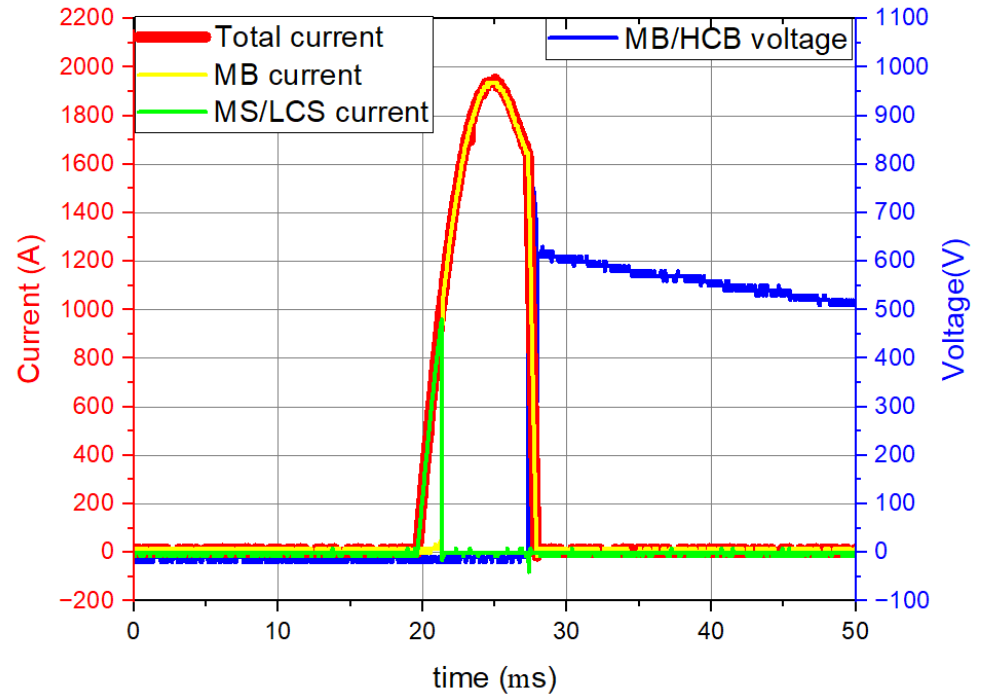
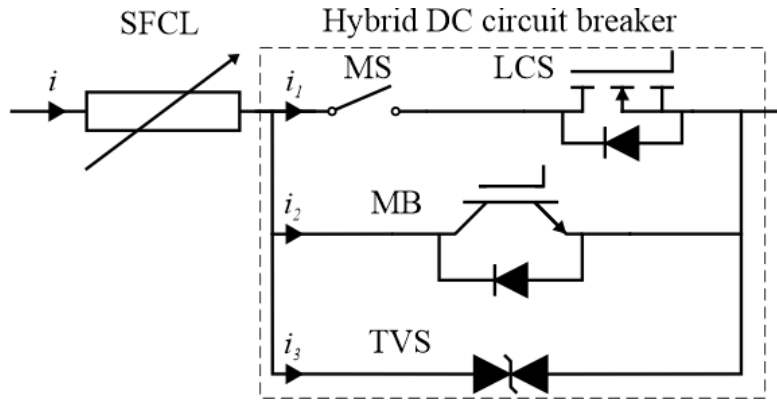
Solid state main breaker



Hybrid Circuit Breaker Testing



Hybrid Circuit Breaker Testing

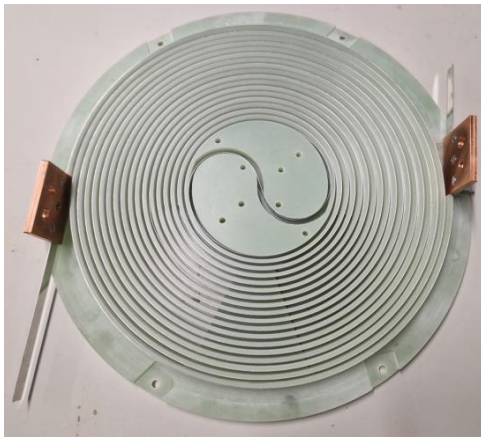


Experimental result

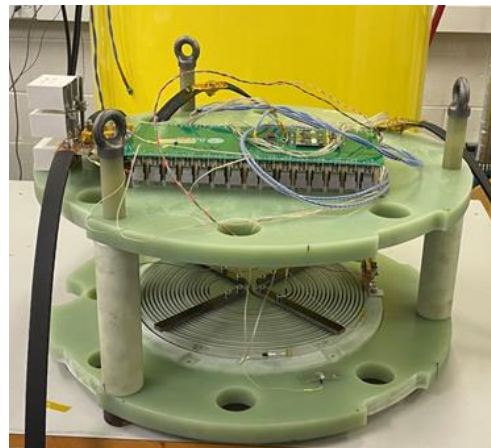
- Hybrid DC circuit breaker interrupts fault current of 1 kA within 5 ms.

SSCB vs HCB

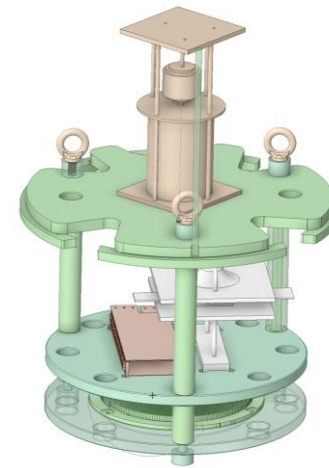
Technical solution	SFCL+SSCB	SFCL + Hybrid CB
Active mass	10 kg	50 kg
Power density	50 kW/kg	10 kW/kg
Loss	500 W	225 W
Efficiency	99.9%	99.95%
Fault clearance time	< 1 ms	5 ms



SFCL



SFCL+SSCB



SFCL+HCB

- Hydrogen electric propulsion address the environment impact and also opens new opportunities for superconductivity technology.
- Superconducting and cryogenic propulsion systems must be considered from an integrated electrical and thermal system perspective.
- Electric aircraft has very high fault current as the on-board power system is closely coupled.
- Superconducting fault current limiter offers effective current limitation allowing DC circuit breakers to operate quickly and reliably, which is a promising solution for on-board DC network protection.
- Integration of SFCL with DC circuit breaker in one cryostat can reduce the thermal leak and on-state losses.



Acknowledgement

- Airbus ASCEND
- UK ATI ZEST 1 - Zero Emission for Sustainable Transport 1
- UK EPSRC Open Fellowship - Towards Zero Emissions Electric Aircraft through Superconducting DC Distribution Network

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