BUILD YOUR OWN AIRCRAFT IN XPLANE 11- PLANEMAKER

BY DAVE ROTHWELL



THIS DOCUMENT IS IN "BETA" – PLEASE FEEL FREE TO SUGGEST IMPROVEMENTS

BUILD YOUR FIRST AIRCRAFT IN PLANEMAKER

SIMPLY FOLLOW THE INFORMATION ON THE IMAGES TO REPRODUCE THE DEFAULT CESSNA 172 IN XPLANE. THE INTERFACE IS SLIGHLTY "UNWELCOMING" AT FIRST BUT YOU WILL GET THE HANG OF IT. FOLLOWING THIS EXAMPLE YOU SHOULD BE ABLE TO GRASP ALL THE CONCEPTS YOU NEED TO DESIGN A COMPLETELY NEW AIRCRAFT FROM SCRATCH.

REMEMBER TO SAVE OFTEN ..



PLEASE DO NOT EDIT ANY DEFAULT AIRCRAFT: LETS MAKE ONE YOU CAN HAVE FUN WITH

You might need to ZOOM IN on this document to see some of the details

CTRL and ROLL your middle mouse button to ZOOM IN.

Planemaker can be found in the XPlane default folder. On Uni PC's that is on the C: Drive

Open Planemaker IMPORTANT - - START A NEW PROJECT - FILE/NEW and SAVE AS - -

in **A NEW FOLDER** in the **EXTRA AIRCRAFT** folder le: C:\X-Plane 11\Aircraft\Extra Aircraft\ **YOURFOLDER** (call it something unique to you)

OPEN THE AUTHOR TAB RENAME YOUR PLANE

×			Author	×
name for		athing		
X-Plane UI	Call you plane some	etning		
call-sign for ATC	Name		1	
tail number for ATC	Give it a Number		Supports user flight	
ICAO code for ATC	[Supports Al flight	
			🗖 Always use Experimental Flight Model	
aircraft author	YOU	· · · · · · · · · · · · · · · · · · ·		
file version				
design studio				
aircraft description	· .			
notes	[
	📄 Ultralight	🗖 Glider		
	Experimental	📄 Seaplane	manufacturer Cessna 🔹	
	🗹 General Aviation	📄 Helicopter	Name yours and give yourself a cal	Isian
	🗖 Airliner	VTOL	Name yours and give yourself a cat	isign
	Military	Science Fiction	Copy the green ticks	
	Cargo			

Your plane will ONLY be saved in that folder on that PC. If you want to keep it for another machine you will have to copy it across – SIMPLY COPY AND PASTE YOUR FOLDER

THERE ARE LIMITED UNDO FUNCTIONS SO SAVE OFTEN IF YOU MESS UP REOPEN THE MODEL WITHOUT SAVING

Press **SPACEBAR** to toggle between SOLID and WIREFRAME model use **WSAD** keys to rotate your model

DON'T BE put off by the complicated interface... you'll get used to it – CLICK THE STANDARD TAB to access the screens you see below you. This first bit is the hardest....

Build the **fuselage** first. Enter the numbers exactly as drag the nodes (dots) into place to copy the Cessna shape. A bit like moulding clay/carving wood USE THE COPY AND PASTE FUNCTION after getting a section right... SAVE IT then work on/reshape the next section. DONT press those silver L/R arrows. that ADDS a section. The numbers ABOVE the sections mark how the location of the fuselage segment. .take your TIME 🙄 get one right, copy it, change it, rpt..

Click the numbers and type in the values - or use the grey up and down buttons...

ZOOM IN TO SAVE YOUR EYES



WINGS

This first wing section is where the FLAPS are situated – green ticks

×							×
Wing 1 Wi	ng 2 Wing	3 Wing 4	Horiz Stab	Vert Stab 1	Vert Stab 2	Wing Flex	snap to RIGT WING 2 🔹
You can c semi-length 0 root chord 0 tip chord 0	Sick the null 0000 (wing ser 0000 THE 25% 0000 (rt) 0000 (rt) 0000 (rt)	FOIL SPECS mbers and t ni-length, root to tij CHORD, not span (sweep 00.0 00.0 (d o 0 1.0	h en type in t) ^{ALONG} long rg) la ver	the box arm 0.03.20 0.03.20 0.03.20 t arm 0.03.20] (ft) texture] (ft) left] (ft) texture	aircraft te top 0.0.2.2 (ratio, to 0.0.0.3 0.0.2.4 [igh 0.0.0.3 0.0.2.4 [igh 0.0.0.4 (ratio, to 0.0.0.4 (ratio, to 0.0.0	cond for this ture part (ratio. bottom p texture top 0.0 2.2 (ratio. bottom cond 1.0 2.2 (ratio. bottom) cond 1.0 (ratio. bottom) cond 1.0 (ratio. b
# 10 incidence ailer roll spoil drag ru elevat elevat elevat rudd rudd rudd rudd rudd rudd rudd rud	ELEM 0 3.0 0.3.0 0 3.0 0.3.0 0 3.0 0.3.0 0 3.0 0.3.0 0 3.0 0.3.0 0 3.0 0.3.0 0 3.0 0.3.0 0 1 1 0 7 1 0 7 1 1 1 1 1 2 1 1 2 1 1 2 1 1 2 1 1 2 1 1 3 1 1 4 1 1 2 1 1 3 1 1 4 1 1 2 1 1 3 1 1 4 1 1 2 1 1 3 1 1 4 1 1 5 1 1 6 1 1 7 1 1 8 1 1 9 1 1 10 1					pic	of default cessna

The second is where the ailerons are

Wing 1 Wing 2 Wing 3 Wing 4 Horiz Stab Vert Stab	Vert Stab 2 Wing Flex snap to RIGT WING 1 +
FOIL SPECS	
	use second 🗖 for this
(wing semi-length, root to tip, ALONG	aircraft texture — part
semi-length 009,50 THE 25% CHORD, not span (ft)) long arm 003.2	(ft) texture top 0.022 (data top texture top 0.022 (data bottom)
root chord 0 5.2 0 (ft) sweep 0 2.5 (deg) lat arm 0 0 000	0 (ft) left 0.0 0 3 0.0 2 2 (ratio) left 0.0 0 3 0.0 2 (ratio) regime (ratio)
tip chord 003.6 (ft) dihedral 0000 vert arm 002.3	(ft) texture bot 0.006 (ratio, top texture bot 0.006 of wing) (ratio, bottom of wing)
ELEMENT SPECS (ROOT AT LEFT, TIP AT RIGHT)	ustomize hords
AA 000 000 000 000 000 000 000 000 000	
99, 999	99 · 999 ·
alleron 1 V V V V V V	
roll spoiler 1	
drag rudder	
elevator 2 0 0 0 0 0 0 0 0 0 0	
rudder 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	
slat 2 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	
speed brake 2	
incidence with all 2	
incidence with elevtr 1 8 8 8 8 8 8 8	
Incidence with rudder 1	
incidence with vector	
incidence with trim	

Wing 3 is the STRUT under the wings – they join on to Fuselage

WING 3

					wing				X
Wing 1	Wing 2	Wing 3	Wing 4	Horiz Stab	Vert Stab 1	Vert Stab 2	Wing Flex		snap to RIGT WING 1 🜻
			FOIL SPECS				ai	use second	d 🗖 for this
semi-leng	th 0 0 7.5 0 (W	ing semi-lengt IE 25% CHORE	h, root to tip, Al), not span (ft))	.ONG long	g arm 0 0 1.6 0) (ft) textu	re top 0.0 0 0	(ratio, top of wing)	texture top 0.0 0 0 (ratio, bottom of wing)
root choi	d 0 0 0.5 0 (ft) sweet	0 7.0 (deg)	la	t arm 0 0 1.8 0) (ft) let	t 0.000 0.0	00 right (ratio)	left 0.0 0 0 0.0 0 0 right (ratio)
tip cho	d 0 0 0.5 0 (ft) dihedral		ver	t arm - 0 0 1.5 0) (ft) textu	re bot 0.0 0 0	(ratio, top of wing)	texture bot 0,0 0 0 (ratio, bottom of wing)
		ELEMENT SP	ECS (ROOT AT L	EFT, TIP AT RI	GHT) 🗖 🖁	ustomize hords			
# 10 in	ncidence 0 0.0								
rc rc d	aileron 1 = aileron 2 = Ill spoiler 1 = rag rudder = elevator 1 = elevator 2 = rudder 1 = rudder 2 =						I		
spe spe incidenc incidence w incidence w	flap 1 = flap 2 = slat 1 = ed brake 1 = e with ail 1 = e with ail 2 = ith elevtr 1 = ith elevtr 2 =								
incidence wil incidence wil incidence incidence	th rudder 1 📄 th rudder 2 📄 with vector 📄 e with trim 📄								

There is no wing 4..

×					Wings	5				×
Wing 1	Wing 2	Wing 3	Wing 4	Horiz Stab	Vert Stab 1	Vert	Stab 2 Wing	Flex	snap to RIGT WI	NG 1 ≑
		F	OIL SPECS					use secor aircraft textu	nd 🗖 for this	
semi-length		(wing semi-lengt) THE 25% CHORD	h, root to tip, AL , not span (ft))	ONG long a	rm 0 0 0.0 0] (ft)	texture top	0.000 (ratio, top of wing)	texture top 0.	(ratio, bottom of wing)
root chord		(ft) sweep	0 0.0 (deg)	lat a	rm 0 0 0.0 0) (ft)	left 0.0 0	0.000 right (ratio)	left 0.0 0 0	0.0 0 0 (ratio)
tip chord		(ft) dihedral		vert a	rm 0 0 0.0 0	(ft)	texture bot	0.000 (ratio, top of wing)	texture bot 0.	(ratio, bottom of wing)
		ELEMENT SP	ECS (ROOT AT LI	EFT, TIP AT RIGH	τ) 🗖 G	istomize oords				

HORIZONTAL STAB.

×							X
Wing 1 Wing 2	Wing 3	Wing 4	Horiz Stab	Vert Stab 1	Vert Stab 2	Wing Flex	snap to RIGT WING 1 🜻
semi-length 0 0 5.3 0	F (wing semi-length THF 25% CHORD	OIL SPECS	ONG long	arm 0 1 7.6 0	(ft) texture	use seco aircraft textu chchch top 0.0 2 4 of wind	nd for this part for this texture top 0.024 (ratio, bottom
root chord 0 0 4.5 0	(ft) sweep (ft) dihedral	0 5.0 (deg)	lat vert	arm 0 0 0.2 0	(ft) left	0000 0000 right 0000 0000 right 0000 0000 ratio, top	left 0.003 0.025 right texture bot 0.004 fratio
	ELEMENT SPE		EFT, TIP AT RIG	HTT)	istomize lords		0000
Image: Second Secon							

×										
Wing 1	Wing 2	Wing 3	Wing 4	Horiz Stab	Vert Stab 1	Vert	Stab 2	Wing Flex		snap to RIGT WING 1
		F	OIL SPECS							
								airc	use second	for this
semi-len	gth 0 0 5.9 5 (wi	ng semi-lengt	h, root to tip,	ALONG Ion	00000	(ft)	texture to	0.0 2 2 (r	atio, top	texture top 0.0 2 2 (ratio, bottom
	00000	L 25 % CHORD		,	00000		A		wing)	
root cho	ord 0 0 5.0 0 (ft)	sweep	3 9.0 (deg	I) la	at arm 0 0 0.0 0] (ft)	left 0.	.0 0 6 0.0 2	4 right (ratio)	left 0.0 0 6 0.0 2 4 (ratio)
tip cho	0.0.2.3.5 (ff)	dihedral	<u>0900</u>	(RIGHT wing)	0.0.0.3.0] (ft)	texture bo	0.0 0 7 (r)	atio, top	texture bot 0.0.0.7 (ratio, bottom
	00000		9999 O	(LEFT wing)	00000			0000.01	wing)	OOOO of wing)
		ELEMENT SP	ECS (ROOT AT	LEFT, TIP AT R	IGHT) 🗖 ct	istomize lords				
# 08	incidence 0 0.3	0 0.2 0 0.1			0.0 0 0.0					
00	aileron 1 🗖									
, r	aileron 2 🗖 oll spoiler 1 📄									
r	roll spoiler 2 📑 drag rudder 🧮									
	elevator 1 🗖									
	rudder 1 🗹									A
	flap 1									
	flap 2 🗖 slat 1 🗖									(a a construction of the second se
sp	slat 2 🕅 eed brake 1 📄									
sp inciden	eed brake 2 📄 ce with ail 1 🥅									
inciden	ce with ail 2 🗍 with elevtr 1 🗖									
incidence v	with elevtr 2 📄									
incidence w	ith rudder 2									
incidence	e with vector 🗎 ace with trim 🗐									
						_	_			



Wing flex

×								×
Wing 1	Wing 2 V	Ving 3 Wing 4	Horiz Stab	Vert Stab 1	Vert Stab 2	Wing Flex		
		Wing flex specs.	Remember to enter	all wing geom	netry to match ir	n-flight at gross v	veight at 1 G.	
			wing midpoint dihedral increase	0.0 0 (midp degre	oint dihedral inc es per G of lift a ion of total	rease in above 1-G)		
			wing flex damping	1.0 0 UUUUUUUUUUUUUUUUUUUUUUUUUUUUUUUUUUU	to lt)			

Landing gear.

X		
Gear Loc Gear Data	Gear Cons	
gear type single 🔹	single 🜻	single 🔹 none ᆃ
long arm - 0 0 0.6 0		0 0 4.6 0 (ft)
lat arm 0 0 0 0 0 0		0 0 1.6 0 (ft)
vert arm - 0 0 1.4 2	- 0 0 1.8 5	- 0 0 1.8 5 (ft)
lon angle		005 (deg)
lat angle 0 0 0 extended 0 0 0	- 0 6 2	062 (deg)
lon angle		000 (deg)
lat angle retracted		000 (deg)
eagle-claw, 0 0 0 2.1 leg length 0 0 0 0 2.1		0 0 0 3.0 (deg, ft)
tire radius, semi-width		0.6 5 0.2 3 (ft)
n-w steering, slow and fast		0 0.0 0 0.0 (deg)
retract axis, strut compres		000 00.0 (deg, ft)
cycle time 0 1.0		0 1.0 (sec)
brakes		castors



×	Landing Gear	×
Gear Loc Gear Data	Gear Cons	
gear deflections	DEFLECTIONS	
damping constants	DAMPING 0.5.9.1 0.5.9.1 0.0.00 (lb/(ft/s))	

NOW THE VIEWPORT SECTION IN THE STANDARDS TAB

×	Viewpoint X
General Cockpit Interior Light	s Exterior Lights Dock Ports Bouncers Sliders Arc Colors
Airspeed speed units knots (does not affect 3-D cockpits)	Cockpit cockpit General Aviation Cockpit
Vso 040 at gross Wso weight	
Vs 048 at gross Weight	long arm pilot's 0 0 3.5 0 (ft)
Vy 000 at gross weight	lat arm pilot's 0 0.0.80 (ft) viewpoint
Vfe-m 085 full dep	vert arm pilot's 0 0 1.4 5 (ft) viewpoint 0 0 1.4 5 (ft)
Vfe-1 110 1st det	
Vno 120	
Vne 163	
Mmo 0 0.4 0 (Mach)	Man
pos G 0 4.5 0 2.0 neg G(limit)	map
airspeed indicator shows autopilot airspeed setting	only airports 🔽 🗖 only paved
	min rway length to show on maps 1.000 (rt)



THESE NEXT THREE ARE NOT ESSENTIAL



General	Cockpit	Inte	rior Lights Exterio	r Lights Dock Por	ts Boun	cers SI	iders	Arc Colors			
longiti	udinal, lateral	, and vertic	al object attach arr	m heading and	d pitch offse	ts (deg)					
clear	002.20	- 0 0 8.8 0 - 0 0 8.8 0	0 0 2.7 0 Sizerof	reference 🔹 0		8.0 VV	anding	🔹 🖌 Hide	# 00 #	length 3 0 0 2 0 width	
clear	0 0 2.2 0	00000 008.30		reference		0.0	Taxi	P Hide		rgb 1.00 1.00 1.00 ength 200 30 rgb 1.00 1.00 1.00	
clear	021.50	000.00	0 0 5.9 0 alrerati	reference 0		0.0 Rotati	ng Beacon	🔹 🗸 Hide	# 00	000 000 000	
clear	02340	00000		reference • 0		0.0 N	av Tail	🔹 🗸 Hide			
clear		010.19 00000	0 0 0.3 6 LEE	WING 2 0		0.0 99	av Left	Nide			l
clear		00000 -010.19	0 0 0.3 6 LEE	WING 2 0		0.0 Antico UU	llision Strobe	• 🗸 Hide	# 00 #		1
clear		010.19	0 0 0.3 6 RIGT	WING 2 + 0		0.0 Na	v Right	🔹 🗹 Hide			
Add	00000	00000	00000	۵	000 00	66			66		

General Cockpit	Interior Lights Exterior Lights Dock Por	Bouncers Sliders	Arc Colors	
has push-back attach for ground towing	000.00 000.00 000.00 vertarr	t. n (ft)	has food- truck 1	00000 00000 long, lat. 000000 000000 vert arm (ft)
has towing hook to tow other craft	018.60 00000 00000 00000 00000 00000 00000 0000	t. ň (ft)	has food- truck 2	00000 00000 long. lat. vert arm (ft)
has towing hook to be towed	0 0 0.0 0 0 0 0.0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	t. n (ft)		
has winching hook to be winched	00000 0000 00000 00000 long, lat	t, ha n (ft)	s baggage- truck 1	00000 00000 long, lat, vert arm (ft)
		ha	s baggage- truck 2	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
has aerial 🗖 refueling dispenser	0 0 0.0 0 0 0 0.0 0 0 0 0.0 0 0 0 0 0 0 0 0 0 0	t, ha n (ft)	s baggage- truck 3	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
has aerial 🗖 refueling receiver 🗖	000.00 000.00 000.00 long. lat 00000 00000 000.00 vert arr	t, (ft) ha	s baggage. truck 4	00000 000.00 00000 00000 Vert arm (ft)
has boarding door 1 🗄	00000 00000 00000 long, lat 00000 00000 00000 vert arr	t, n (ft)	has crew- car 1	00000 00000 long, lat, 00000 vert arm (ft)
has boarding door 2 🗖	000.00 000.00 000.00 long, lat 00000 00000 000.00 vert arr	t, (ft)	has crew. car 2	00000 00000 long, lat, vert arm (ft)
has fueling 😿 truck 1	-00000 00000 00000 long, lat 00000 00000 00000 vert arr	t, n (ft)	has ground DODOO	00000 00000 long, lat, 00000 vert arm (ft)
has fueling truck 2	00000 00000 00000 long, lat 000000 00000 00000 vert arr	t, (ft)		

Bouncers tab – just the top line

×			Viewpoint		×
G	eneral Cockpit Interio	or Lights Exterior Lights Dock Ports	Bouncers Sliders	Arc Colors	
	has this 💼 bouncer can redline bouncer 💼 float freely acceler	rotor 0 0.5 0 (G) spring k per uni mass lon, lat, ver	t 0,0 0 0 0,0 0 0,0 0 0,0 0 0	damp k per unit mass lon, lat, vert	01.0 01.0 001.0 001.0 001.0 001.0 001.0 001.0
	has this 🗖 bouncer can redline bouncer 🗖 float freely acceler	rotor 0.0.0 (G) spring k per un ration 00.0 (G) mass lon, lat, ver	t 0,0 0 0 0 0,0 0 0 0,0 0 0	damp k per unit mass lon, lat, vert	00.0 00.0 max travel 00.0 0

X

Sliders and ARC Colours can be ignored

ENGINE SPECS

×	Engines X
Engines 1 Engines 2 Props 1 Pro	os 2 jets 1 jets 2 jets 3 Trans/Radiator Start/Spoolup SFC/Sound
GENERAL ENGINE SPECS	PROP ENGINE SPECS
critical 00,000 (max altitude at which full power avail at zero speed)	maximum 00,180.00 (hp) turbine and recip hp to 000 (lb/hp) redline 02,700 (engine RPM)
FADEC automatically keeps engines from exceeding max allowable power or thrust	RAM-inlet 0.0 0 (fraction) top of 0.6 5 0 (engine RPM)
throttle available at max lever, one engine failed 1.00 UUU	loss of oil press to fine pitch prop, blue knob full aft bottom of green arc bottom of 0 1,9 0 0 0 0,0 (deg) 0 1,9 0 0 0,0 (deg) 0
throttle available at max 1.00 lever, all engines running 000 (throttle)	all engines: have throttle- governor for helos beta pitch of prop, thottle at idle 0000 0000 minimum prop governor RPM FADEC limits RPM on all propeller engines 0000 0000 0000 00000
hi idle fuel adjustment 1.1 8 (fuel ratio)	FADEC sets fuel-air ratio on recip engines reverse pitch of prop, throttle full reverse 000 0.0 (deg) idle 00000 0000 (engine RPM)
lo idle fuel adjustment	 all propellers: feather when prop control at min all propellers: feather when mixture control at min all propellers: feather when all propellers: feather and the regime failure all engines: shut off fuel
go to low afterburner 0.0 0 above this throttle	JET ENGINE SPECS
go to BETA PITCH below this throttle lever position	maximum 000000 (lb) compressor 00000 (square fan RPM at 000000 (rpm) allowable thrust 000,000 (lb) compressor 000000 feet) fan RPM at 000,000 (rpm)
this throttle lever position	afterburner thrust increase (lb) reverser 000.00 (square reverser 000.00 (square reverser)
throttle available at max reverse lever position	ROCKET ENGINE SPECS at sea level, optimum altitude, and vacuum
	thrust 0,0 0 0,0 0 0.0 0,0 0 0.0 0,0 0 0.0 0,0 0 0.0 (ib) nozzle 0 0 0.0 (square exit area 0 0 0.0 (square feet)
	optimum altitude 000000 (feet) SFC 09.9.7 (/hr)
auto-set RPM and throttle based on power lever	



×								×
Engines 1	Engines 2	Props 1	Props 2	Jets 1	Jets 2	Jets 3	Trans/Radiator Start/Spoolup SFC/Sound	
					PROP	SPECS		
# prop # b	lades 02	¢ CW ¢						
	clutched							
prop ra	dius 0 0 3.3 0] (ft)						
tip chord	ellipse 0.0 0 fraction 0.0 0 0.0 0	(inches)						
fine and coarse pitch		(deg)						
desi at pr	ign TAS op disc 110	1						
design A at root and	AOA 0.0 0.0 0.0	(RPM) (deg)						
AOA r spanwi	oot to tip 1.0 se power 🛛	1						
max all	lowable op pitch 9 0.0	1						

Engines 1 Engines 2 Propriat matrix 1 matrix 2 matrix 3 Transchadulation Start/Spaceup StC/Sourd Number 00000 00000	Engines 1 Engines 2 Props 1 Props 2 jets 1 jets 2 jets 3 Trans/Radiator Start/Spoolup SFC/Sound Rib LE 0000 0000 0000 0000 0000 0000 0000 0000 00000 <
Rib.L.E. 0.000	$ \begin{array}{c c c c c c c c c c c c c c c c c c c $
Ð	Element MAA 0.315 0.328 0.352 0.355 0.426 0.472 0.522 0.574 0.629 0.685 Element AOA 0.675 - 1675 - 21.05 - 21.77 - 20.90 - 19.46 - 17.92 - 16.45 - 15.13 - 13.95

SOME OF THESE ABOVE NUMBERS WILL VARY AS YOU PROGRESS IN THE BUILD. THEY MIGHT BE UNEDITABLE AT THIS TIME

~	1	1	-11	1	1	-1	1			~
Engines 1	Engines 2	Props 1	Props 2	Jets 1	Jets 2	Jets 3	Trans/Radia	tor Start/Spoolup	SFC/Sound	
					TRANSISSI	IONS				
transmission	0.0 2 0 (part) numb	er of 1 (#)	engine #: transmi	1 drives	this engine free-wheel	e p Is by	rop #1 is driven		
105505	9999		₽		9				9	
					RADIATOR PER	ENGINE				
	radiator in propwas	is 🗖								
		0000				66	00			
	du expansio	ct 0 0.0 0 (r	atio)		radia area Pi	ER ENGINE	.0 0 (square feet)			
		~~~~				0	66			
					coeffic	radiator 0 cient of drag	0.0 (cd)			







воттом

#### ENGINE NACELLE - TRY THE "ELLIPSE" FUNCTION

SYSTEMS TAB



#### GO TO LIMITS 1 AND 2 TAB

^	3951	ems	^
General 1 General 2 Electrical 1 Electrical 2	Para Bus 2	Limits 1 Limits 2	Warnings
[	REDLINE YELLOW GREEN-AR	RC-RANGE YELLOW REDLINE	
			%)
	N2 000 0 000		%)
primary is 🔴 EPR primary is 💽 N1			ratio to ambient ressure)
fuel flow ⊖ lbs/hr fuel flow ⊙ gal/hr fuel flow			pounds per hour r gallons per hour)
torque Opercent			percent or ft-lb)
power opercent power horsepower			percent or orsepower)
	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	~~~~~	
max MP 02 9.9 0 (max POWER, redline RPM, sea level, ISA)			inches)
max ITT or TIT based on engine type			degrees C or F)
max EGT 1,2 5 0 degrees C degrees F			degrees C or F)
max CHT 0,5 0 0 O degrees C			degrees C or F)
l			

×					X
General 1 General 2 Electrical 1 E	lectrical 2 Bus 1	Bus 2 Limits	1 Limits 2	Warnings	
	REDUNE YELLOW	GREEN ARC RANGE	VELLOW REDUNE		
max oil temp			C 2 4 5 C or F)	es)	
nominal oil 080 (any units)			(any u	nits)	
nominal fuel 0.30.0 (any units)			0 3 0.0 (any u	nits) electric fuel 0.3 (pump press 0.3 (0 0 0 1 0 0 0 1 0 0 0 1 warning
battery voltage 0 2 4.5 (volts)			0 2 4.5 (volts)		
generator voltage		0 2 4.5 0 2 6.5 0 0 0 2 6.5	0 2 6.5 (volts)		
max battery 0,0 6 0 (amps)			(amps)) max output each e	ngine-driven generator
max APU/GPU/ RAT amperage 0.060 (amps)			0,0 6 0 (amps)) 160 (amps)	
nominal xmsn 001.0 (any unit	s)		0 0 0 0.0 (any u	nits)	
nominal xmsn temperature	s)		0 0 0.0 (any u	nits)	
max instrument 05.5 suction 05.5			0 6.0 (any u	nits)	
APU N1 100			(any u	nits)	

WARNINGS TAB IS NOT REQUIRED.

WHO NEEDS WARNINGS EH?

Some of the tabs have been missed out to keep it simple at this stage. I have all the slides if you need to get more complex. . but at this level you won't need that info.

Time for some wings

CONTROL GEOMETRY is next...

This is an important bit

Oh and save regularly .. if you haven't been

SPECIAL/ select show with STILL/MOVING controls – your plane will come to life..

X	X Control Geometry X				
Controls	rim & Speed Pha	ase-Out Speedbrakes Stick Forces			
aileron 1 chord ratio		CONTROL SIZES 15.0 20.0 control surface down then up allerons corrugated with gaps *	FLAP AND SLAT COEFFICIENTS slat 1 Krueger flap from L.E.D. deployment (deg)		
chord ratio	0.0 0 0.0 0	0 0.0 0 0.0 down then up	slat 2 krueger flap from L.E.D. deployment		
elevator 1 chord ratio		15.0 2 0.0 control surface down then up elevators corrugated with gaps +	flap 1 slotted flap flap root flap tip flap tip flap tip 0.2 6 fl		
elevator 2 chord ratio		0 0.0 0 0.0 down then up	flap 2 plain flap flap root chord ratio flap tip flap		
rudder 1 chord ratio		2 0.0 2 0.0 Control surface 0 0 0 0 left then right rudders corrugated with gaps €	flap Cl 0.9 1 4 flap Cd 0.0 5 9 flap Cm 0.2 8 2		
rudder 2 chord ratio		00.0 00.0 control surface 00.0 00.0 left then right	flap Cl 0.000 flap Cd 0.000 flap Cm 0.000		
roll spoiler 1 chord ratio		0.0 roll roll input 0.0 (ratio)	FLAP AND SLAT DEFLECTIONS		
roll spoiler 2 chord ratio		00.0 roll roll input 0.00 (ratio)	flap speed ratio at deploy-start		
drag rudder chord ratio		0.0.0 drag vudder def	flap detents 03. flaps are infinitely-		
speedbrake 1 chord ratio		0 0.0 speedbrake 0 0.0 FLIGHT def			
speedbrake 2 chord ratio		0 0.0 speedbrake			
Plain flaps are t pivots down, pre	the simplest type oducing some ext	of flap. They are simply a control surface that tra lift and drag.	slat 1 0.0 1.0 1.0 000 000 000 000 slat 2 0.0 0.0 0.0		



NO SPEEDBRAKES.. who needs SPEEDBRAKES?

×		Geometry X
Controls Trim & Speed Phase-O	ut Speedbrakes Stick Forces	
none		none
none		none

X Control Geometry X						
Controls Trim & Speed Phase-Out Speedbrakes Stick Forces						
reference speed for pitch force for roll force for for full- right alleron force for full- right rudge force for force for full- right rudge	ND DAMPING Ditch DO 0.0.0 (lb) damping DO 0.0.1 (lb) roll damping DO 0.0.2 (lb) roll damping DO 0.0.2 (lb) roll damping DO 0.0.2 (lb) yaw damping DO 0.0.2 (lb) yaw damping DO 0.0.3 (lb) yaw damping DO 0.0.4 (lb) yaw damping DO					
STALL AND SHAKER STALL AND SHAKER aero stall force output force frequency output force force frequency output force f						
HYDRAULIC FORCE INCREASE force multiplier for flight controls on hydraulic failure force multiplier for brakes on hydraulic failure 0 0 1.0 (ratio)	FORCE MAXIMUM max allowable pitch force pitch force p					

WEIGHT AND BALANCE.



×	Weight & Balance	X
Weight & Bal Tanks CG Set CG Check		
tank #1 ratio 0.5 0 lat tank CG 0000 (ft)	tank #2 ratio 0.500 lat tank CG 002.00 (ft)	tank #3 ratio
full tank 0 0 3.7.0 (ft) tank long 0 0 3.7.0 (ft) full pump 10 (psi) CG location 0 0 2.3.5 (ft)	full tank 0 0 3.7.0 (ft) empty tank long 0 0 3.7.0 (ft) full pump 10 (psi) vert tank 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	fuel pump 10 pressure 00 (psi)
Role Normal Tomoves with NONE	Role Normal 🔹 moves with NONE 🜩	Role Normal 🕈
tank #4 ratio 0.0 0 0	tank #5 ratio	tank #6 ratio
fuel pump 10 (psi) pressure 0 (psi)	fuel pump 10 (psi) pressure 10 (psi) Role Normal 1	fuel pump 10 (psi) pressure 00
tank #7 ratio	tank #8 ratio	tank #9 ratio
fuel pump 10 (psi) pressure 00 (psi) Role Normal T	fuel pump 10 (psi) pressure 0 (psi) Role Normal 0	fuel pump 10 (psi) pressure U

×		Weight & Balance	×
Weight & Bal Tanks	CG Set		
	Enter the arm and maximum weight (Then go to the next tab over to see th	of the passengers and cargo that you expect to carry here. e range of weight and balance you could encounter in flight!	
1	NOTE: THIS IS ONLY FOR WEIGHT AND BALANCE I	NVESTIGATION FOR YOU, AND DOES NOT AFFECT X-PLANE IN ANY WAY!	
0 0,0 0 1,7 2 1.0 (lb)	0 0 3.3 0 (ft) empty craft		
66666666	66666	6666666666666666	
0 0,0 0 0,1 7 5.0 (lb)	003.70 (ft) fuel tank #1	0 0,0 0 0,0 0 0.0 (lb) 0 0 0.0 0 (ft)	
	000370 (ft) fueltank #2		
	00000 (ft) fueltank #2		
	00000 (ft) fuel tank #4		
	00000 (ft) fuel tank #5		
	00000 (ft) fuel tank #6		
9999999999			
0.0.0 0.0.0 0.0 (lb)	000.00 (ft) fuel tank #7		
999999999	99999	00000000 00000	
0 0.0 0 0.0 0 0.0 (lb)	0 0 0 0 0 (ft) fuel tank #8		
999999999	99999	00000000 00000	
0 0,0 0 0,0 0 0.0 (lb)	0 0 0.0 0 (ft) fuel tank #9		
00000000	99999	0000000 00000	



Go to the EXPERT TAB - select AIRFOIL

You only need the first two ta	bs – props and wings
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×					Foil	5		×
Props	Wings	Misc Wings	Misc Wings	Misc Wings	Misc Wings	Pylons 1	Pylons 2	
lo Re 📄	Clark-Y (root).afl Clark-Y (root).afl			Clar	Prop 'k-Y (good prope 'k-Y (good prope	1 :ller).afl :ller).afl		span interp
					Prop	2		
					Prop	3		
-					Prop	4		
					FIOP	~		
					Prop	5		
					Prop	6		
					Prop	7		
					Prop	8		
					Prop			

X	Foils	×
Props Wings Misc Wings Misc Wings	Misc Wings Misc Wings Pylons 1 Pylons 2	
lo Re 📄 NACA 2412-root.afl hi Re 📄 NACA 2412-root.afl	WING 1 NACA 2412-root afl	variable-sweep variable-dihedral variable-incidence retractable
lo Re TNACA 2412-root.afi hi Re NACA 2412-root.afi	WING 2 NACA 2412-tip.afl	variable-sweep variable-dihedral span interp 0.00 variable-incidence power 0.00 variable-incidence variable var
lo Re 📄 NACA 0024 (symmetrical).afl hi Re 📄 NACA 0024 (symmetrical).afl	WING 3 NACA 0024 (symmetrical).afl NACA 0024 (symmetrical).afl	variable-sweep variable-dihedral span interp 0.00 variable-incidence power 0.00 variable-incidence 0.00 variable.
	WING 4	
lo Re 📄 NACA 0006 (symmetrical).afi hi Re 📄 NACA 0006 (symmetrical).afi	HSTAB NACA 0006 (symmetrical).afl	variable-sweep variable-dihedral variable-incidence retractable
lo Re NACA 0009 (symmetrical).afl hi Re NACA 0009 (symmetrical).afl	VSTAB 1 NACA 0009 (symmetrical).afl NACA 0009 (symmetrical).afl	variable-sweep variable-dihedral variable-incidence power 1.0 0 power 1.0 0
	VSTAB 2	

BACK THIS PLANE UP!! BEFORE YOU ALTER IT/ CHANGE ITS PARAMETERS.

Save it one last time and open XPlane. Select new flight and try out your creation

Would be best to see if it works as you expected. you might have to tweak a few things but if you want to TOTALLY alter some of the settings, SAVE AS and make A COPY of your aircraft. . changing the AUTHOR settings etc. you can then alter power/cg/wings etc

HAPPY EXPERIMENTING – Dave Rothwell